Experimental Study of a Shock-accelerated Water Layer



With imaging and velocity measurement

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Introduction

• A large vertical shock tube is used to experimentally investigate a flat water layer subjected to a shock wave.

• The shock wave accelerates the shocked liquid layer down the shock tube where it is imaged in the test section to study the shock induced breakup.

• Pressure history along the tube and the endwall are measured to study shock mitigation



PT4

PT3

-PT2

-PT1

PTB

Imaging Techniques Shadowgraphy

- Nd:YAG laser (532 nm), 10 ns pulse.
- Beam expanded and collimated to 25 cm diameter.
- Image focused on semi-transparent screen and recorded with a scientific grade high resolution CCD camera.



and impulsive force due to the water layer.

The Shock Tube

- 9.2 m long, vertically oriented.
- Square internal cross section, 25 x 25 cm.
- Up to Mach 5 into atmospheric air.
- Water layer supported by 1 μ m Mylar film.

High-speed Cinematography



- High-speed digital camera at 30,000 fps
- 128 x 256 pixel resolution

Inter face

Test Section

Basement

- Back lighting with continuous white light source
- Field of view: 3.8 x 7.6 cm
- Image sequence from M=3.04 shock onto 12.8 mm thick water layer









• M=2.68 in argon. • initially 6.4 mm thick of water.

1.47 ms

• Time measured from initial shockwater contact.

• A projection (2nd derivative of density) of the entire shock tube cross section, including the shock wave, is visible with this technique.

Planar Imaging

- Nd:YAG laser (532 nm), 10 ns pulse.
- Expanded to a laser sheet with 15° divergence and 1 mm thickness.
- Mie scattering/absorption by water.
- Cross sectional image recorded with scientific grade CCD camera.







• M=2.14 in argon. • Initially 12.8 mm thick of water. • Time measured from initial shockwater contact.



Velocity Measurement: away from the end-wall

- Photodiode array: 12 sensors, 1.27 cm apart.
- Backlight through semi-transparent screen.





Velocity Measurement: near the end-wall





• A false wall inserted on the bottom flange to determine velocity on impact.

• 6 blue light sensitive photodiodes, 1.17 cm apart.

• Illuminated by a collimated continuous argon-ion laser beam (mixed wavelength 457.9-514.5 nm).





Future Work

• Tests at M=2.14 and M=3.20, 46 cm below initial position.

• The leading edge of the water travels at nearly constant speed (shocked-gas particle velocity ±12%) regardless of the initial thickness.

• Water layer decelerates as it approaches the end-wall.



• The leading edge of the water travels 65% more slowly than the particle velocity for a 6.4 mm layer and 52% more slowly for the 12.8 mm layer due to the reflected shock.

• Due to interaction with reflected shock, water layer slows down before reaching the end-wall.

End-wall Impulse

- Pressure measurements at the end-wall are integrated over time.
- The impulse is compared to a test without water at the same Mach number.
- At M=1.38, the impulse decreases
- At higher M, the impulse increases by up to 90% due to the water layer.



Since the main body of the water layer is not visible in either shadowgraphy or planar imaging, X-ray radiography will be used to measure the volume fraction of the layer. This will yield a quantitative measure of the mixing within the layer. A new interface section of the shock tube will accommodate several layers of water for studying the mitigating effects that multiple layers may provide compared with a single layer.