



**Progress Report on Work Performed on the LIBRA  
Design During the Period 1 February 1987 to 31  
January 1988**

**B. Badger, G. Moses, R. Engelstad, G. Kulcinski, E.  
Lovell, J. MacFarlane, Z. Musicki, R. Peterson, M.  
Sawan, I. Sviatoslavsky, L. Wittenberg**

**January 1988**

**UWFDM-754**

***FUSION TECHNOLOGY INSTITUTE  
UNIVERSITY OF WISCONSIN  
MADISON WISCONSIN***

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***Agenda for LIBRA Review Meeting,  
28–29 January 1988***

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<b>Channel Formation</b>	<b>Peterson</b>
<b>30 MV Accelerator Design</b>	<b>Smith</b>
<b>Target Injection</b>	<b>Peterson</b>
<b>Cavity Clearing</b>	<b>Peterson/Moses</b>
<b>Neutronics</b>	<b>Moses</b>
<b>Cavity Design</b>	<b>Kulcinski</b>
<b>Tritium Retention</b>	<b>Kulcinski</b>

## *Plasma Channels are Required for Repetitive Light Ion Fusion*

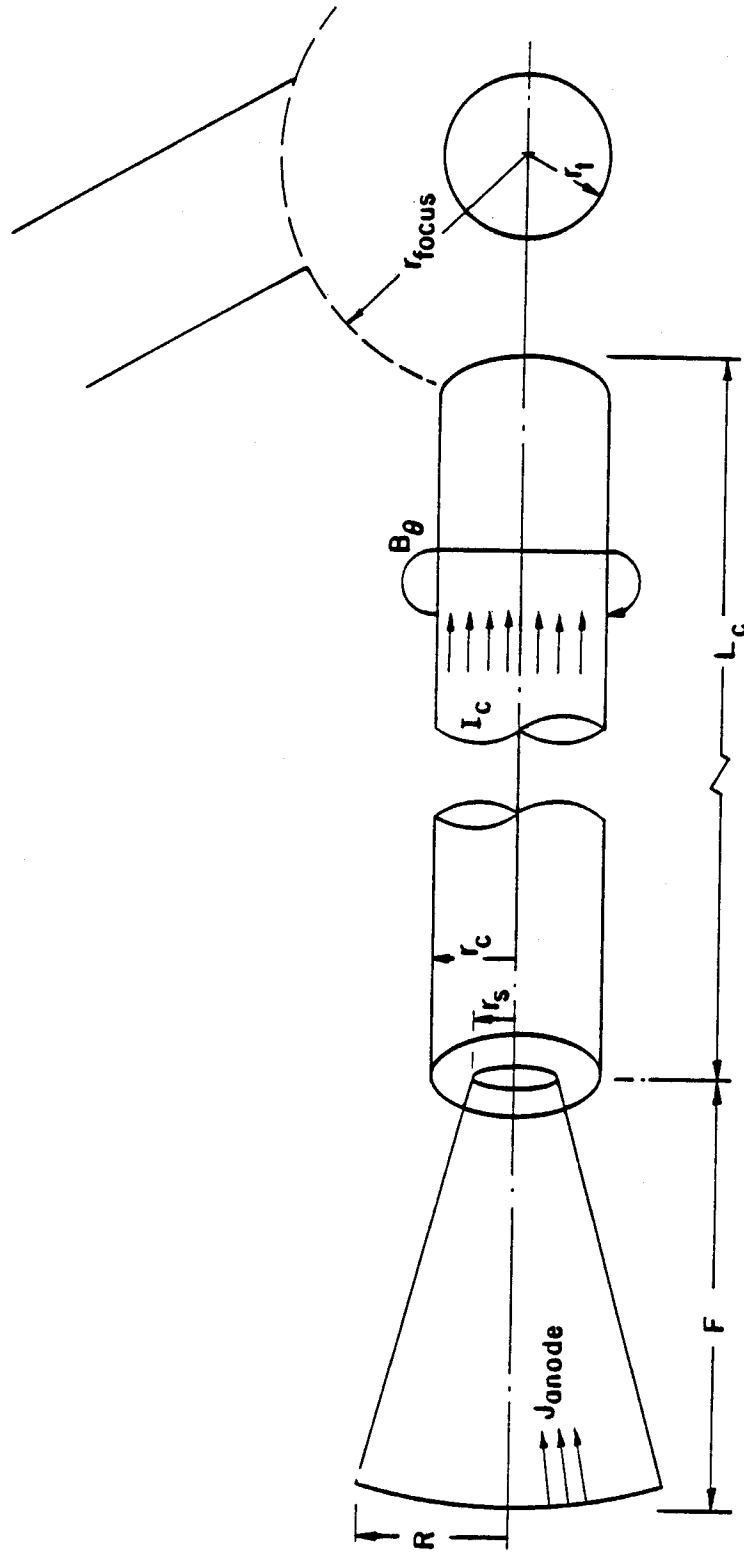
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**Purpose of Channels:** To carry beams of ions from ion diodes to a target efficiently and repetitively, to provide standoff for ion diode protection, and to allow time-of-flight beam bunching.

**How They Work:** A discharge electron current forms azimuthal magnetic fields that confine ions to the channel and rarifies the channel center to minimize collisional ion energy loss.

**How a Channel is Formed:** Two lasers preionize narrow tubes of the background gas that intersect at the target, then a high voltage capacitor is discharged across the ends of these tubes.

# SCHEMATIC OF PLASMA CHANNEL



DIODE	DRIFT REGION	CHANNEL ENTRANCE	CHANNEL	FOCUSSING REGION	TARGET
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## ***Channel Parameters for LIBRA***

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Channel Length	5.4 m
Channel Radius	0.5 cm
Target Chamber Gas	$3.55 \times 10^{17} \text{ cm}^{-3}$ Ar + 0.2% Li
Average Beam Ion Energy	30 MeV
Maximum Injection Angle into Channel	0.15 radians
Ion Species	$\text{Li}^{+3}$
Ion Power Injected into Channel	9.5 TW
Bunching in Channel	4.4
Ion Power on Target per Channel	25 TW
Ion Energy Loss in Channel	25%
Fraction of Ions Reaching Target	80%
Number of Main Pulse Beams	16
Number of Prepulse Beams	2
Number of Channels	36
Maximum Discharge Current	100 kA
Discharge Current Shape	Double Pulse
Required Magnetic Field at Channel Radius	28 kG

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## *Comparison Between LIBRA Channels and Channels in Other Facilities*

	EAGLE	TDF	LIBRA
Length (m)	4.9	3	5.4
Radius (cm)	0.63	0.5	0.5
Ion Species	D	Li	Li
Ion Energy (MeV)	6.3	30	30
Injection Angle (rad)	0.09	0.15	0.15
Background Gas	He+1% Xe	N <sub>2</sub>	Ar+0.2% Li
Gas Density (cm <sup>-3</sup> )	2x10 <sup>19</sup>	1x10 <sup>18</sup>	3.55x10 <sup>17</sup>
Max. Discharge Current (kA)	30	100	100
Ion Energy Loss (%)	10	25	25
Number of Channels	46	24	36



# *Computer Simulations of Plasma Channel Formation*

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## **ZPINCH Computer Code:**

- 1-D (radial)
- Lagrangian hydrodynamics
- MHD
- Magnetic Field Diffusion
- Current Profile and History
- Radiation Diffusion
- Tabulated LTE equation-of-state

## **Optimization of Channel Parameters:**

- Done for  $N_2$  (TDF)
- Vary laser width
- Vary current history

## **Channel Formation Calculations for LIBRA:**

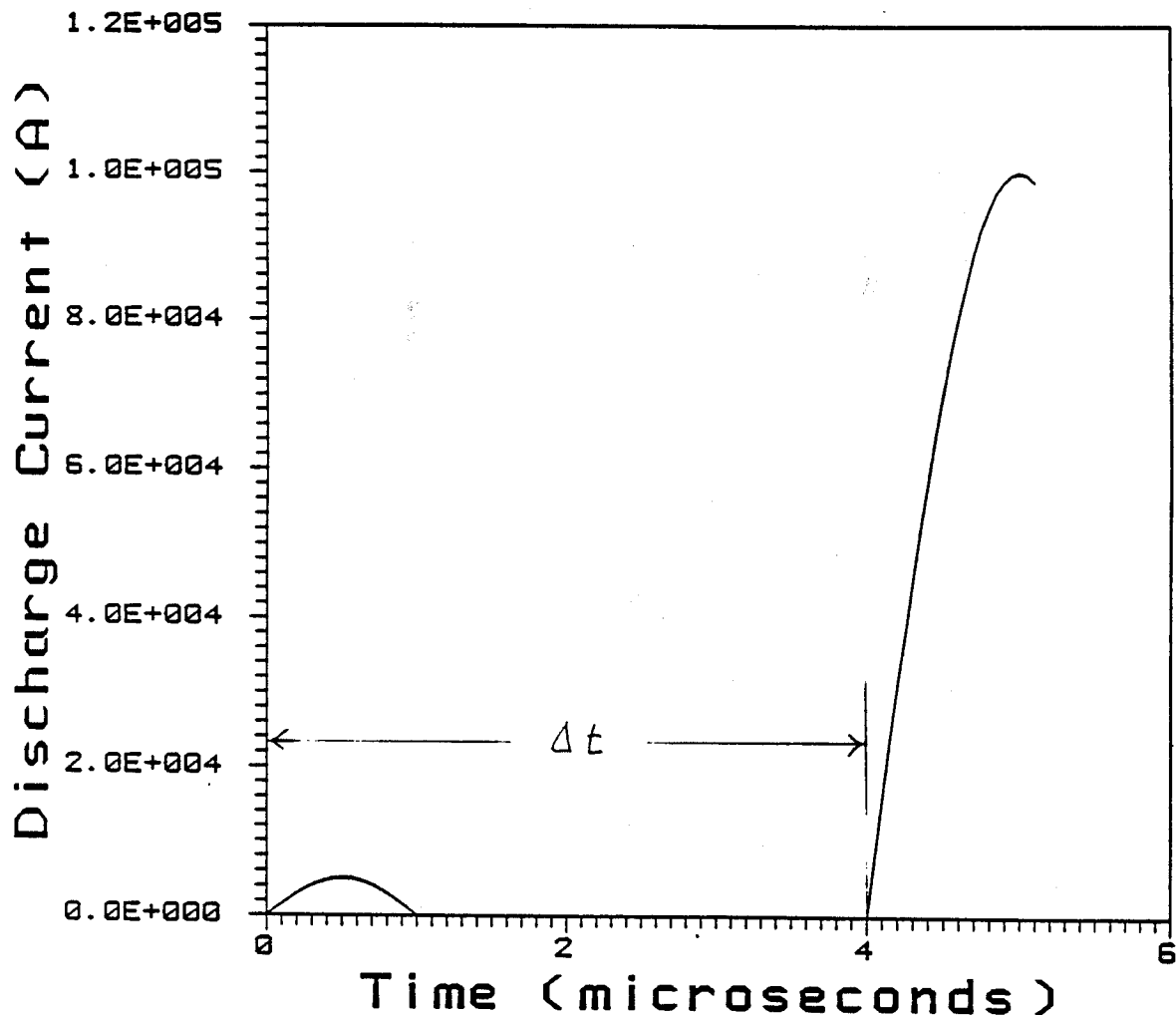
- Done for Ar + 0.2% Li
- Done for best laser and current parameters  
in  $N_2$  optimization
- Done with and without radiation transport

## Double Pulse Channel Discharge Current Profile

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First pulse initiates radial shock that reduces gas density in channel.

Second pulse creates azimuthal magnetic field, but can pinch channel.

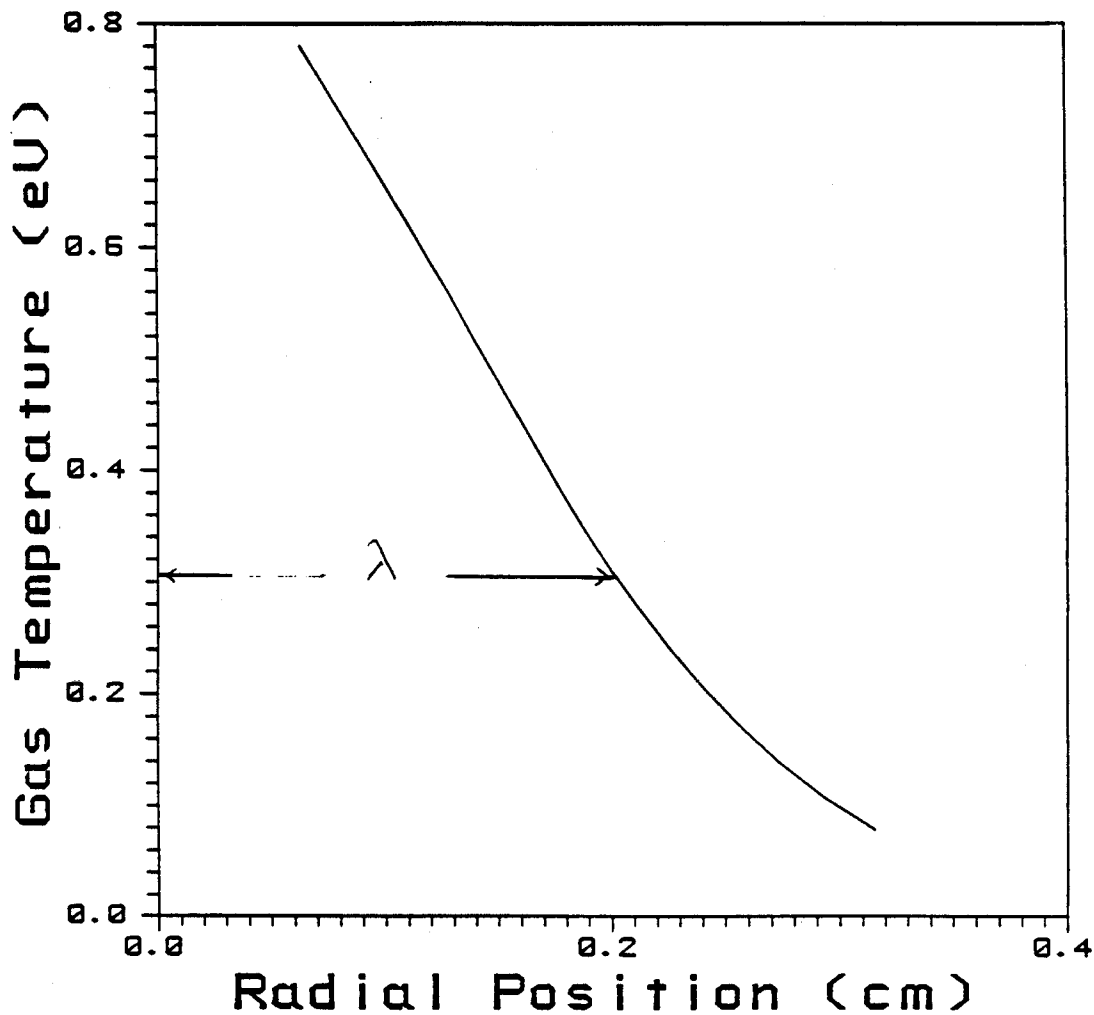


Time delay  $\Delta t$  is an independent variable in ZPINCH input.

## *Laser Profile is Gaussian*

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Gaussian laser profile leads to a Gaussian initial gas temperature profile in the channel.

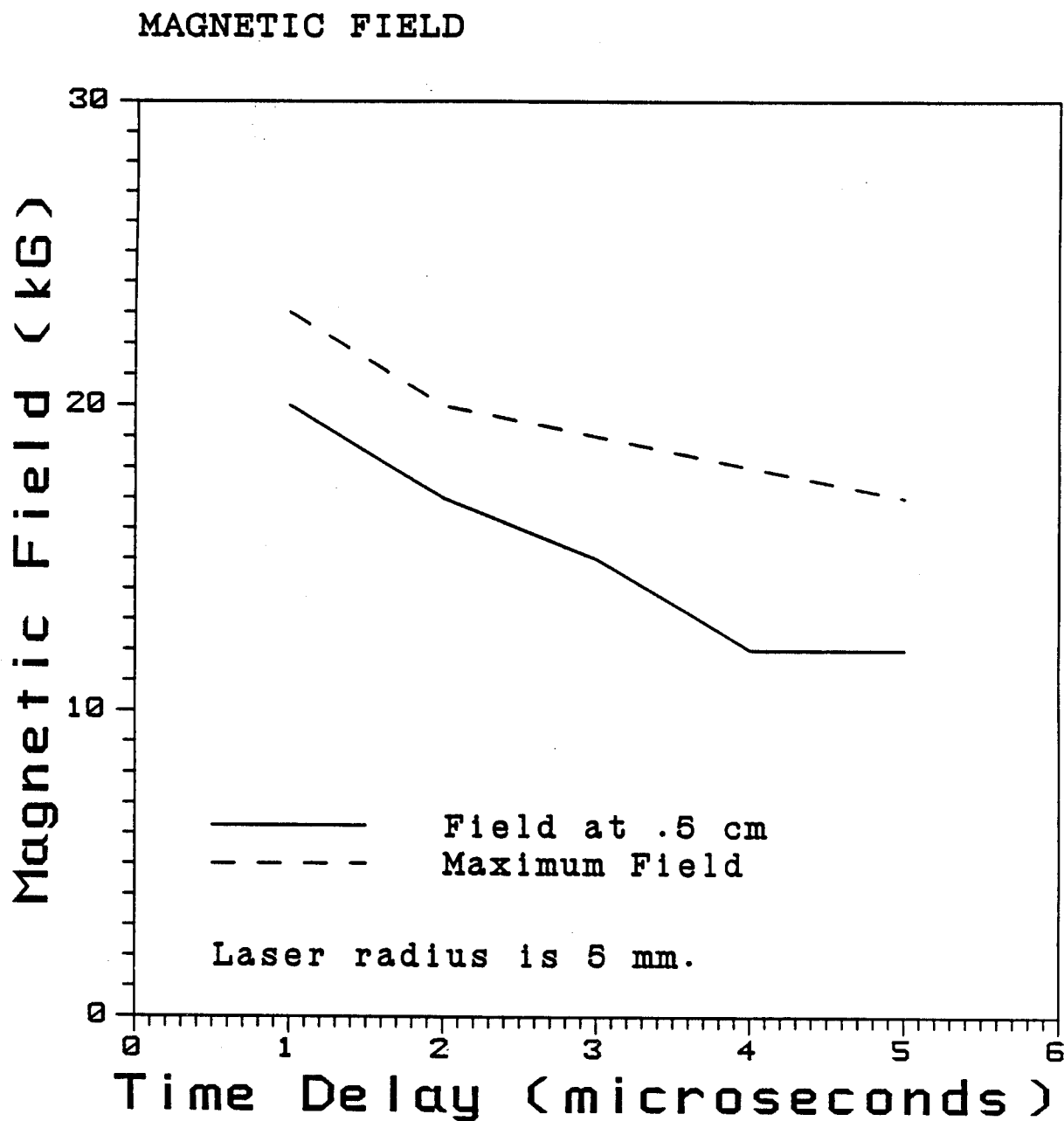


Gaussian width is an independent variable in ZPINCH input.

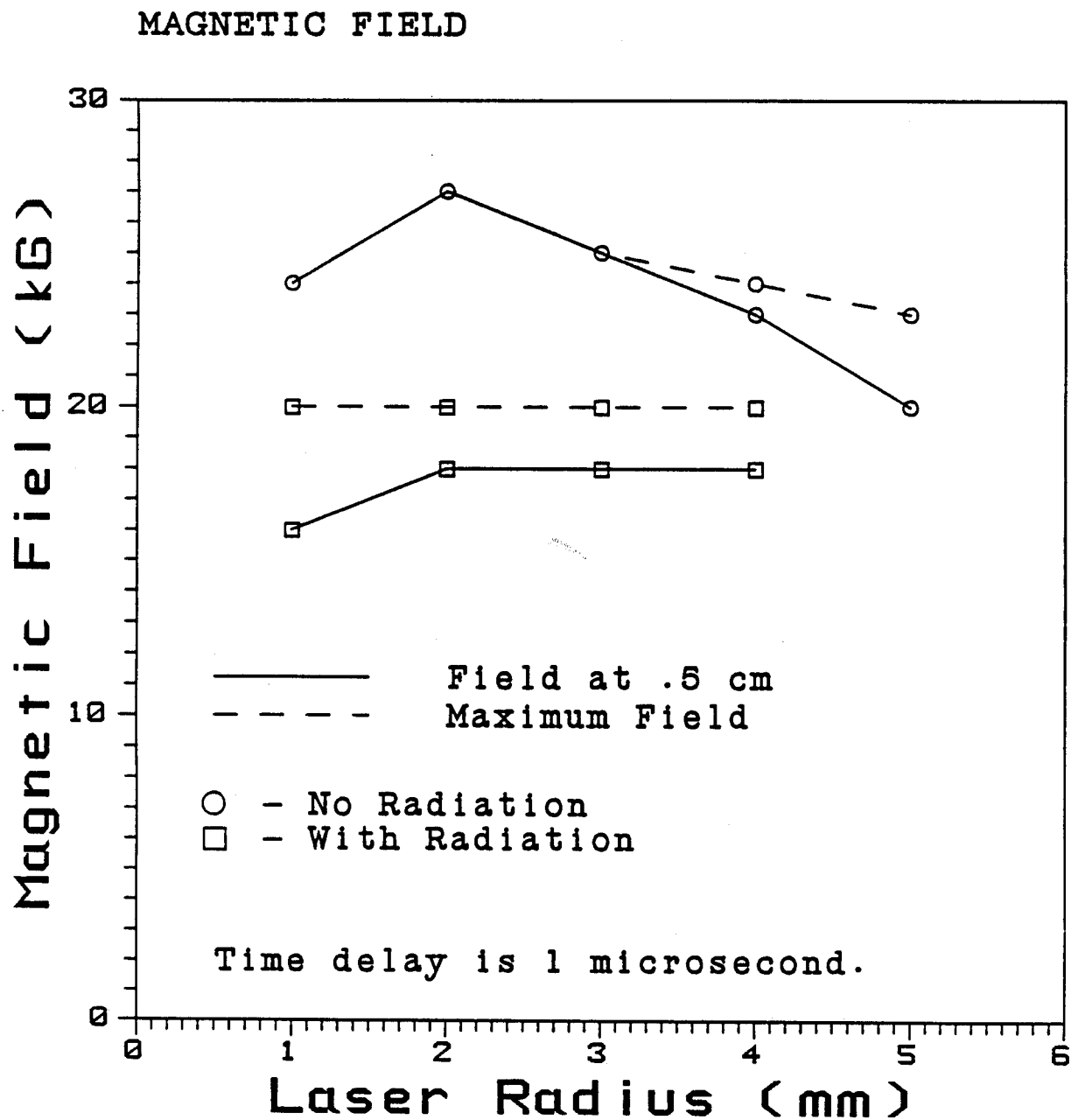
***Matrix of Z-PINCH Calculations***

		Time Delay ( $\mu$ s)				
		1	2	3	4	5
Laser Radius (mm)	1	X	X			
	2	X				
	3	X	X			
	4	X				
	5	X	X	X	X	X

# PARAMETRIC STUDY FOR N<sub>2</sub> CHANNELS



# PARAMETRIC STUDY FOR N<sub>2</sub> CHANNELS

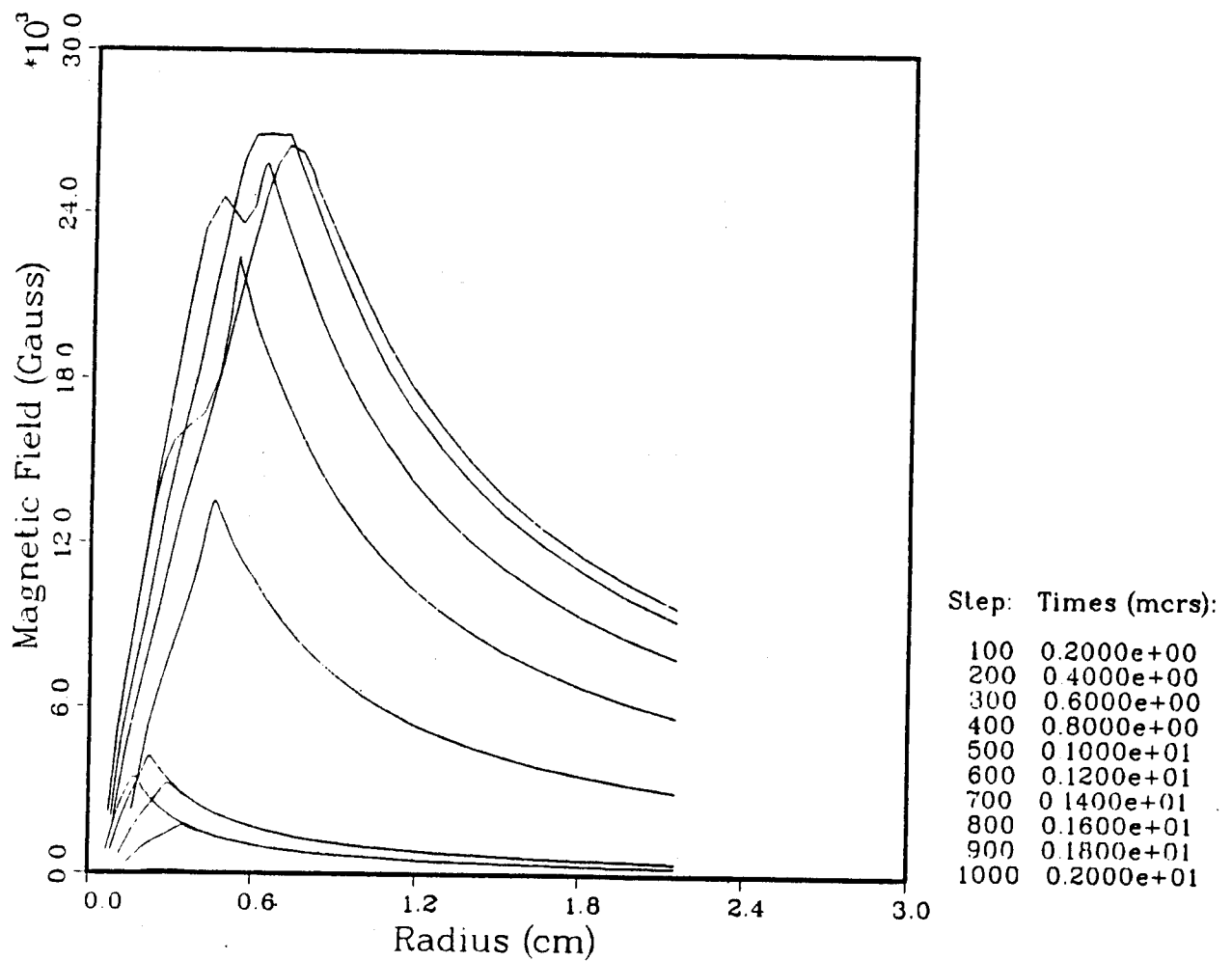


# BEST FREE STANDING CHANNEL-NO RADIATION

$\Delta T = 1 \mu s$ , Laser Width = 2 mm

Gas =  $10^{18} \text{ cm}^{-3} \text{ N}_2$

Magnetic Field vs. Radius

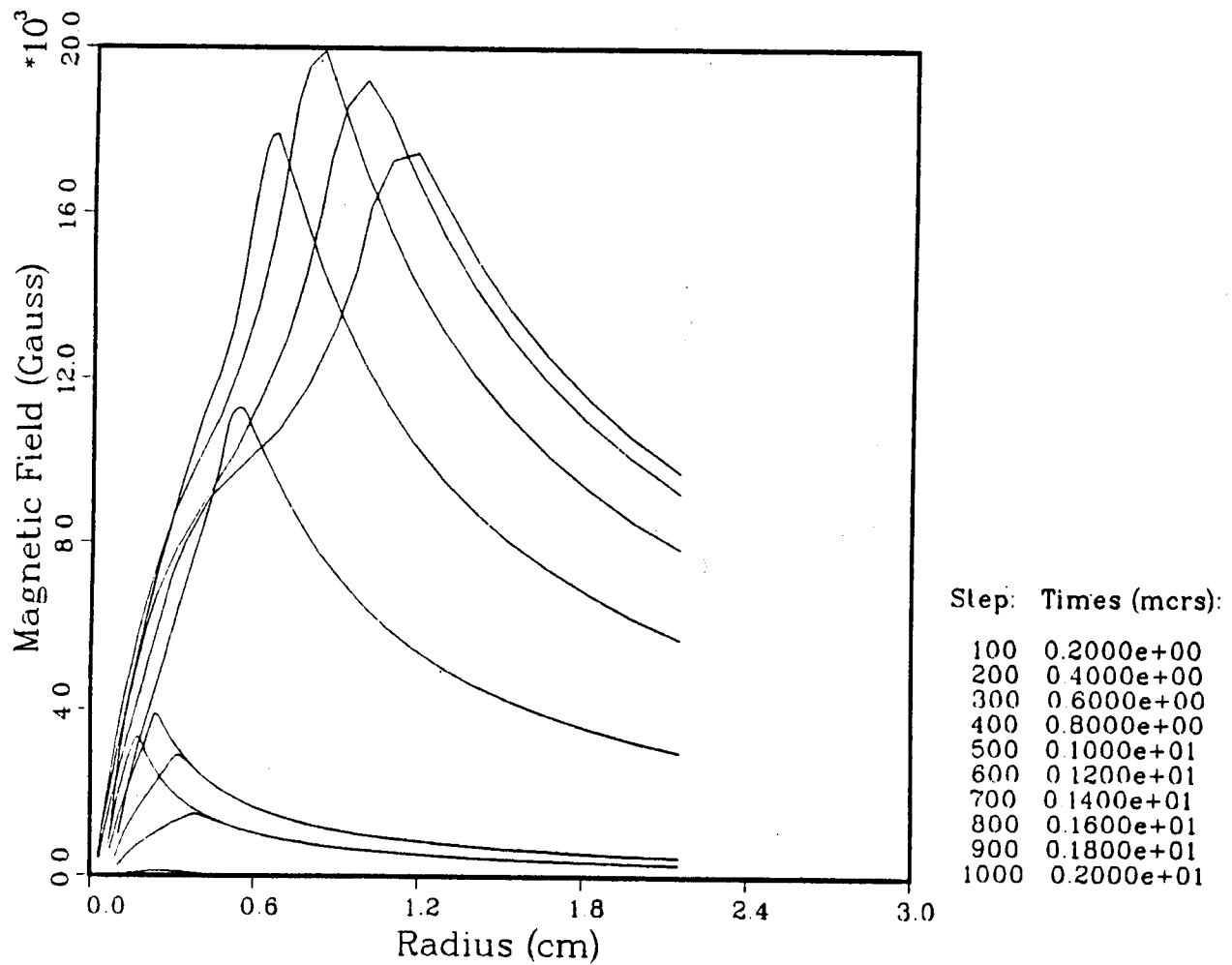


# BEST FREE STANDING CHANNEL WITH RADIATION

$\Delta T = 1 \mu s$ , Laser Width = 2 mm

Gas =  $10^{18} \text{ cm}^{-3} \text{ N}_2$

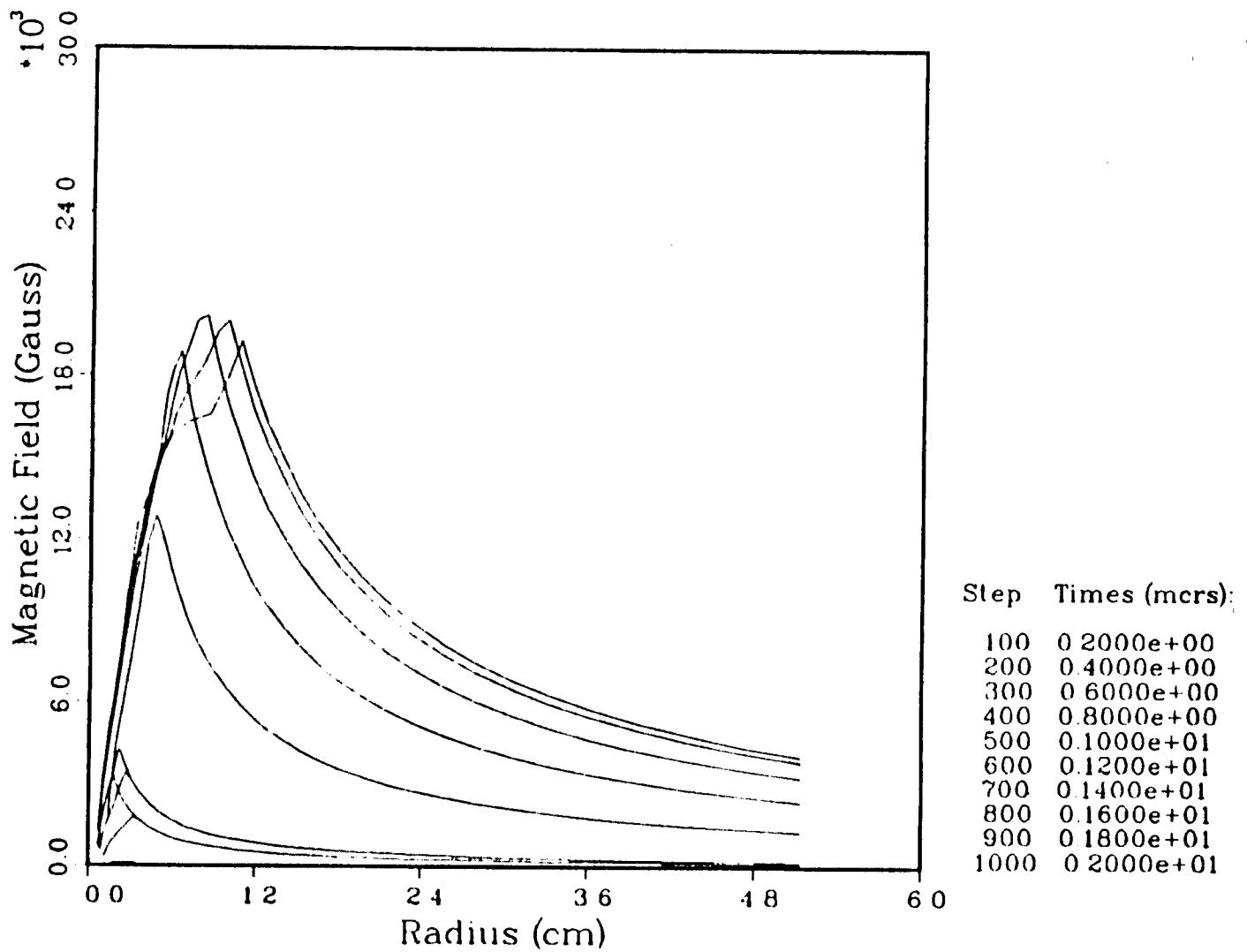
Magnetic Field vs. Radius





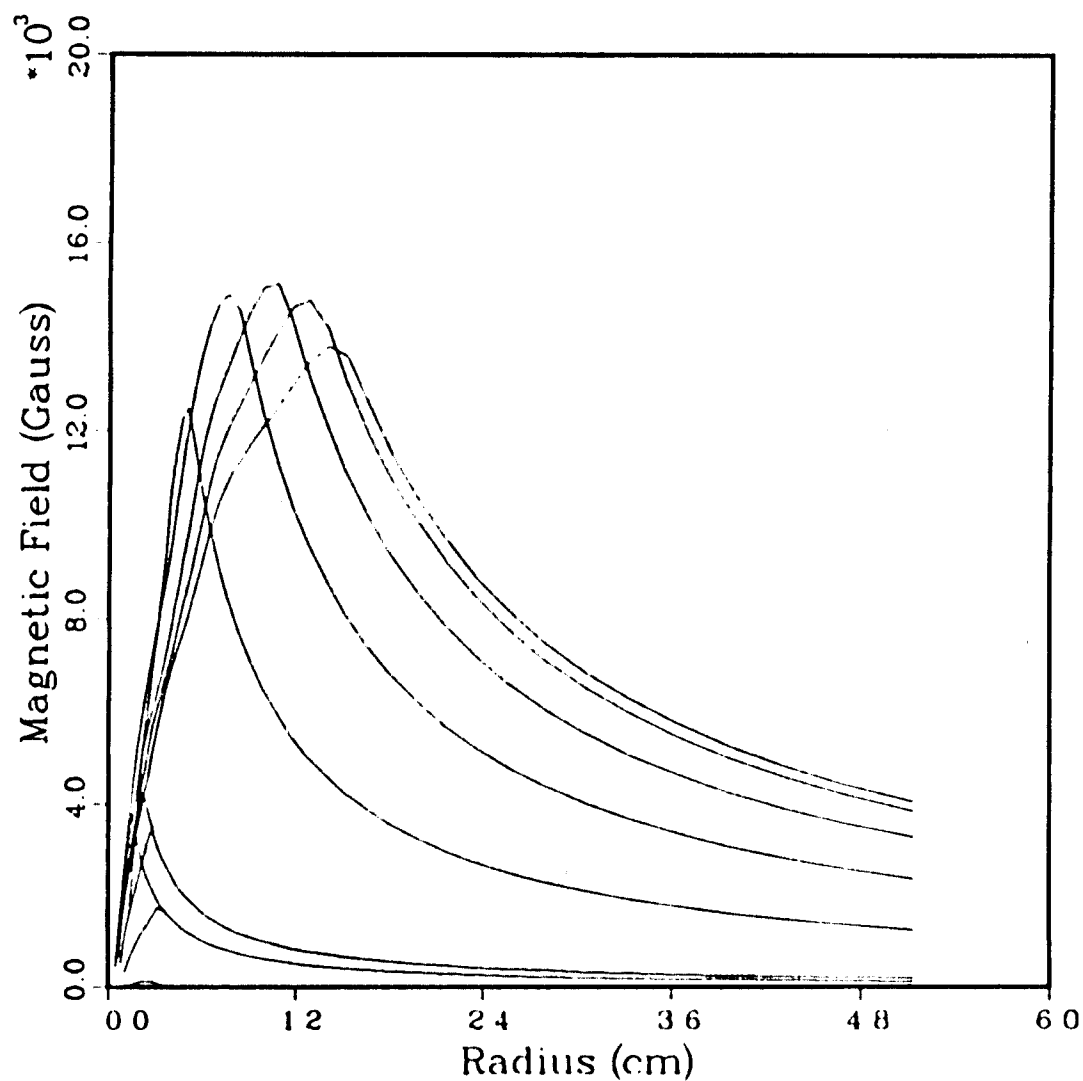
## BEST CASE CHANNEL WITHOUT RADIATION

$\Delta T = 1 \mu s$ , Laser Width = 2 mm  
Gas =  $3.55 \times 10^{17} \text{ cm}^{-3}$  Ar + 0.2% Li



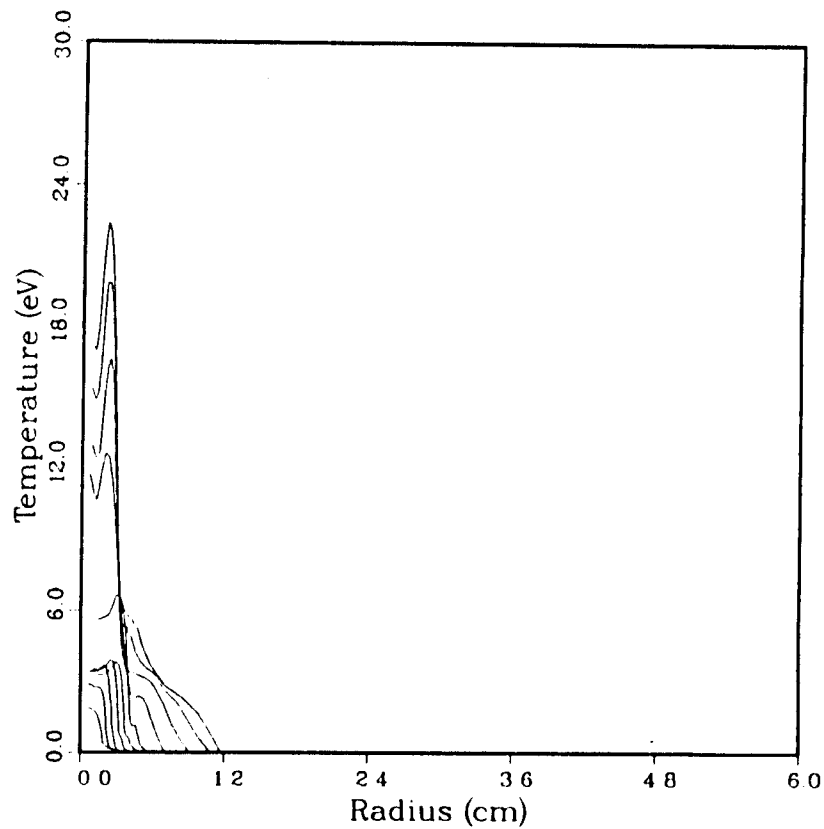
# BEST CASE LIBRA CHANNEL WITH RADIATION

$\Delta T = 1 \mu s$ , Laser Width = 2 mm  
Gas =  $3.55 \times 10^{17} \text{ cm}^{-3}$  Ar + 0.2% Li

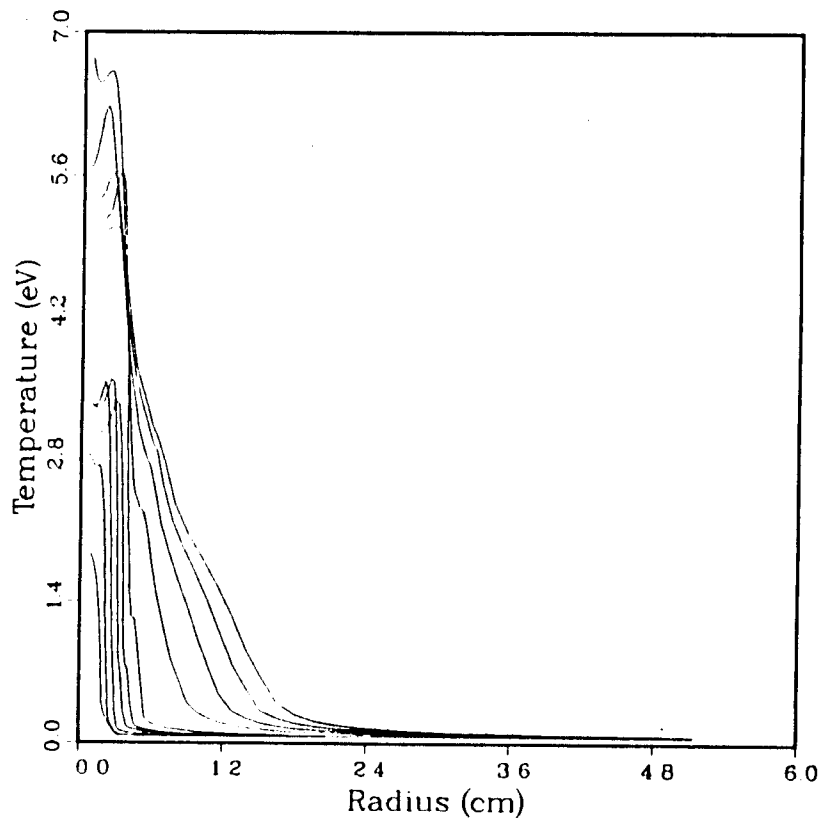


# TEMPERATURE PROFILES IN LIBRA CHANNELS

**Without Radiant  
Heat Transfer**



**With Radiant  
Heat Transfer**



## *Radiation Transport is Important Issue for LIBRA Channels*

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- Radiant heat transfer can spread out hot region of channel, where electrical conductivity is high.
- This spreading causes discharge current to flow in a cylinder of larger radius than if there were no radiation.
- Any current that flows more than 0.5 cm from the channel axis is “wasted” because it only creates magnetic field more than 0.5 cm from the channel axis.
- Therefore, radiant heat transfer can reduce the magnetic field 0.5 cm from the channel axis, which is the place that the amplitude of magnetic field needs to be high.
- From comparisons of radiant heat transfer, as calculated with the radiation diffusion method, with small blast experiments we believe that ZPINCH might overestimate the radiant heat transfer in channels.

## *Is the Current History Realistic?*

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- Inductance of channel is at least  $1 \mu\text{H}$  per meter of channel length

$$L_{\text{channel}} = 2 \times 5.4 \text{ meters} \times 1 \mu\text{H/m}$$

$$= 10.8 \mu\text{H}$$

- The maximum current and risetime of the second discharge current pulse leads to  $\dot{I} \simeq 10^{11} \text{ A/s}$ .

- Therefore the voltage required to drive this current pulse is

$$V_D = L\dot{I} = 1 \text{ MV}$$

- Can such a large electrical potential only cause a discharge down the length of the channel and not to some other structure? We do not know.

## *Conclusions*

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We have discussed:

- 1) Channel parameters for LIBRA and have compared them with EAGLE and TDF
- 2) Optimization of channels in  $N_2$  gas (relevant to TDF)
- 3) Channel simulations in  $Ar + 0.2\% Li$  for LIBRA
- 4) The importance of radiation in channels
- 5) The problems of providing the required current history.

The present status of channels in LIBRA is:

- 1) We have a design that provides a 20 kG magnetic field 0.5 cm from the channel axis, if radiation can be neglected. This should be compared to a design goal of 28 kG.
- 2) If our radiative diffusion calculation is correct we have only 14 kG.
- 3) We don't yet know what problems the high discharge voltage will cause.

Recommendations:

- 1) We should continue optimizing the channel design—but with a new version of ZPINCH that may better predict the radiation.
- 2) The high discharge voltage problem must be considered.

**30 MV Accelerator Design**  
**(viewgraphs available under separate cover)**





## *Heating of the Target During Injection*

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- The cryogenic DT fuel must be in a symmetric hollow shell at liquid or solid density when the driver fires.
- Convective and radiant heat from the target chamber and frictional heating by the injector gun barrel can warm the target.
- We have calculated the temperature in target with a finite-difference heat transfer computer code with temperature-dependent thermal properties.
- We will present:
  - 1) Limits on the fuel temperature
  - 2) Heating of the target during acceleration in gun
  - 3) Heating of the target while in target chamber
  - 4) Conclusions

## ***Limits on Fuel Temperature in Cryogenic Targets***

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- Solid DT fuel

$$T_{\text{fuel}} < 19.7 \text{ K (triple point for DT)}$$

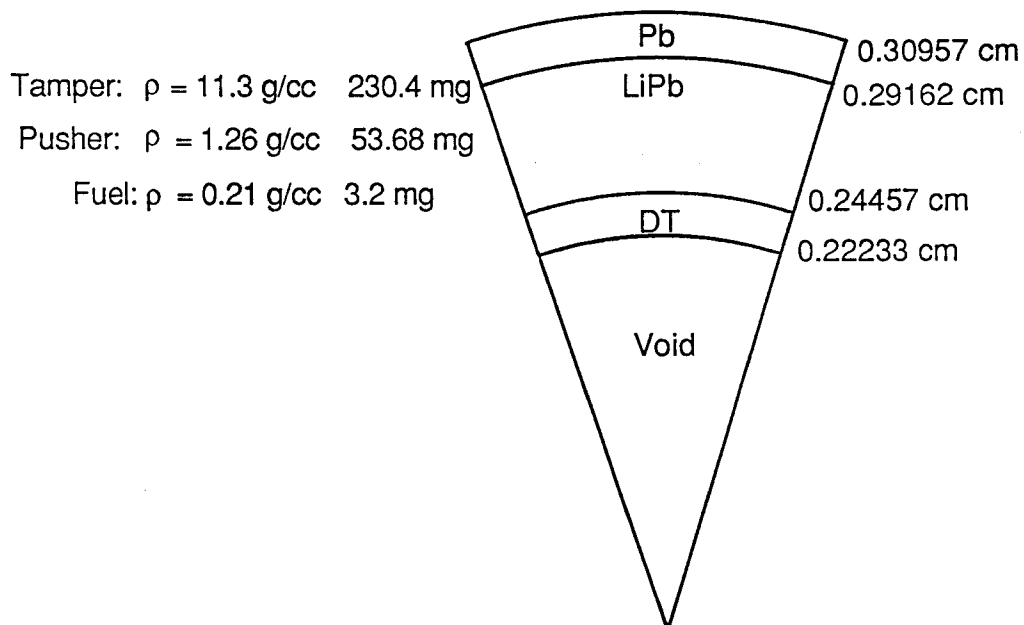
- Liquid DT held in place by a low density rigid foam

$T_{\text{fuel}}$  can be as high as 30 K, depending on the acceptable DT vapor density in capsule center.

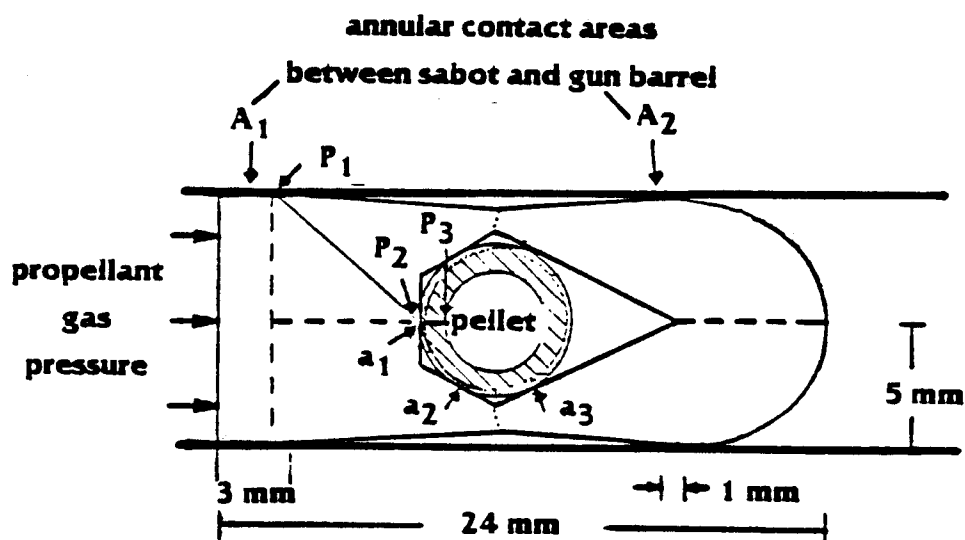
## *LIBRA Target is in Sabot in Gun Barrel but Bare in Target Chamber*

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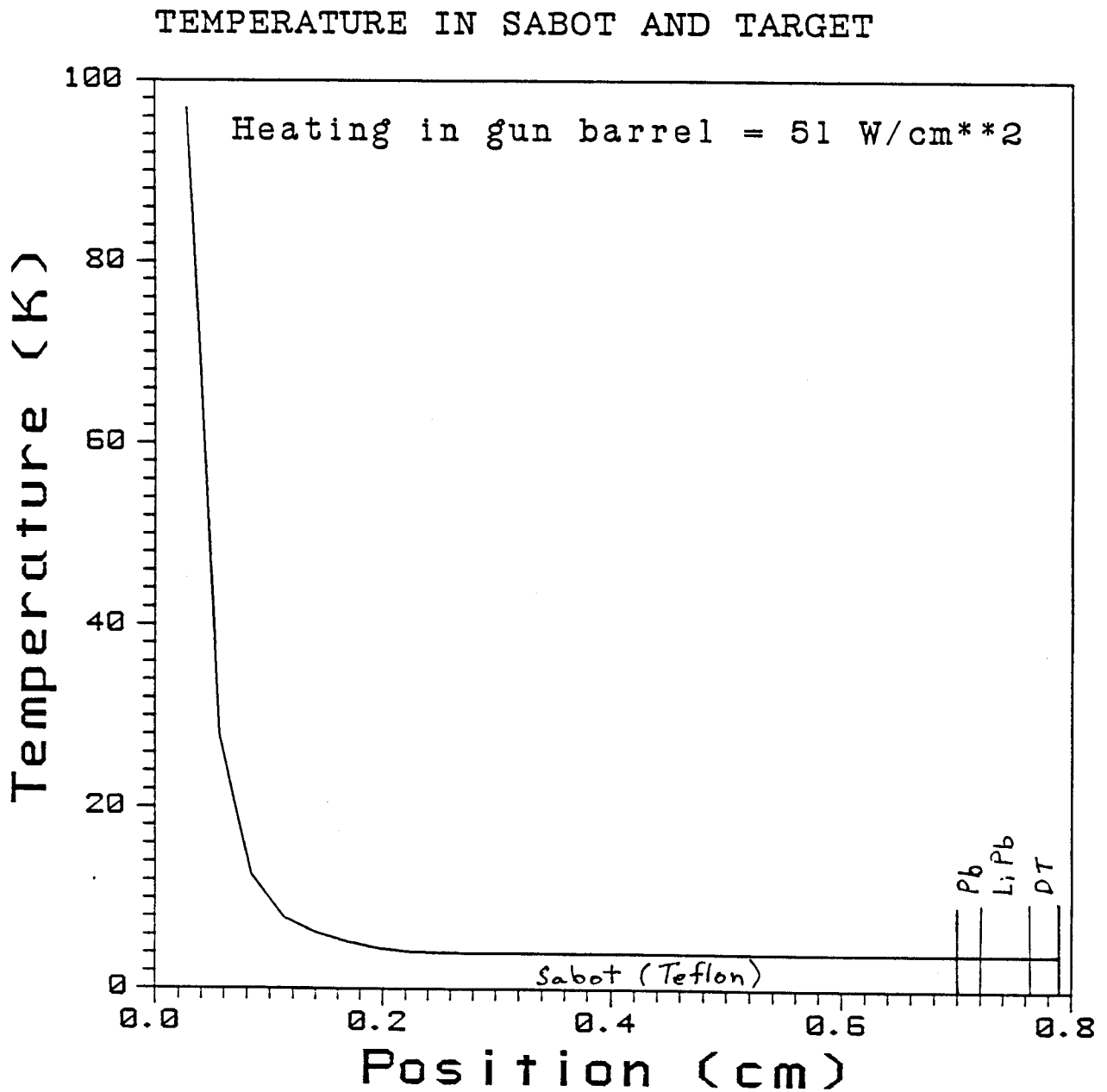
### LIBRA Target Moving Through Target Chamber Gas



### Target in Sabot in Gun Barrel



# Sabot Protects Target from Heating During Acceleration



## ***Heat Loads in Target Chamber Gas***

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### **Convective Heat Transfer**

Parameters	$T_{\text{gas}} = 800 \text{ K}$
	$V_{\text{target}} = 200 \text{ m/s}$
	$d_{\text{target}} = 1 \text{ cm}$
	$\rho_{\text{gas}} = 2.4 \times 10^{-5} \text{ g/cm}^3$
	$t_{\text{injection}} = .025 \text{ s}$
	$\mu_{\text{gas}} = 480 \times 10^{-6} \text{ g/cm-s}$

$$\text{Re} = \frac{V d \rho}{\mu} = 985$$

$$\text{Nu} = \frac{h c d}{k_g} = 0.37 (\text{Re})^{0.6} \quad (\text{Kreith}) \quad = 23$$

$$k_f = \text{thermal conductivity of gas} = 1.9 \times 10^{-4} \text{ W/cm-K}$$

$$h_c = \text{surface conductance} = 4.3 \times 10^{-3} \text{ W/cm}^2\text{-K}$$

$$\ddot{q}_{\text{conv}} \simeq h_c T_{\text{gas}} = 3.5 \text{ W/cm}^2$$

### **Radiative Heat Transfer**

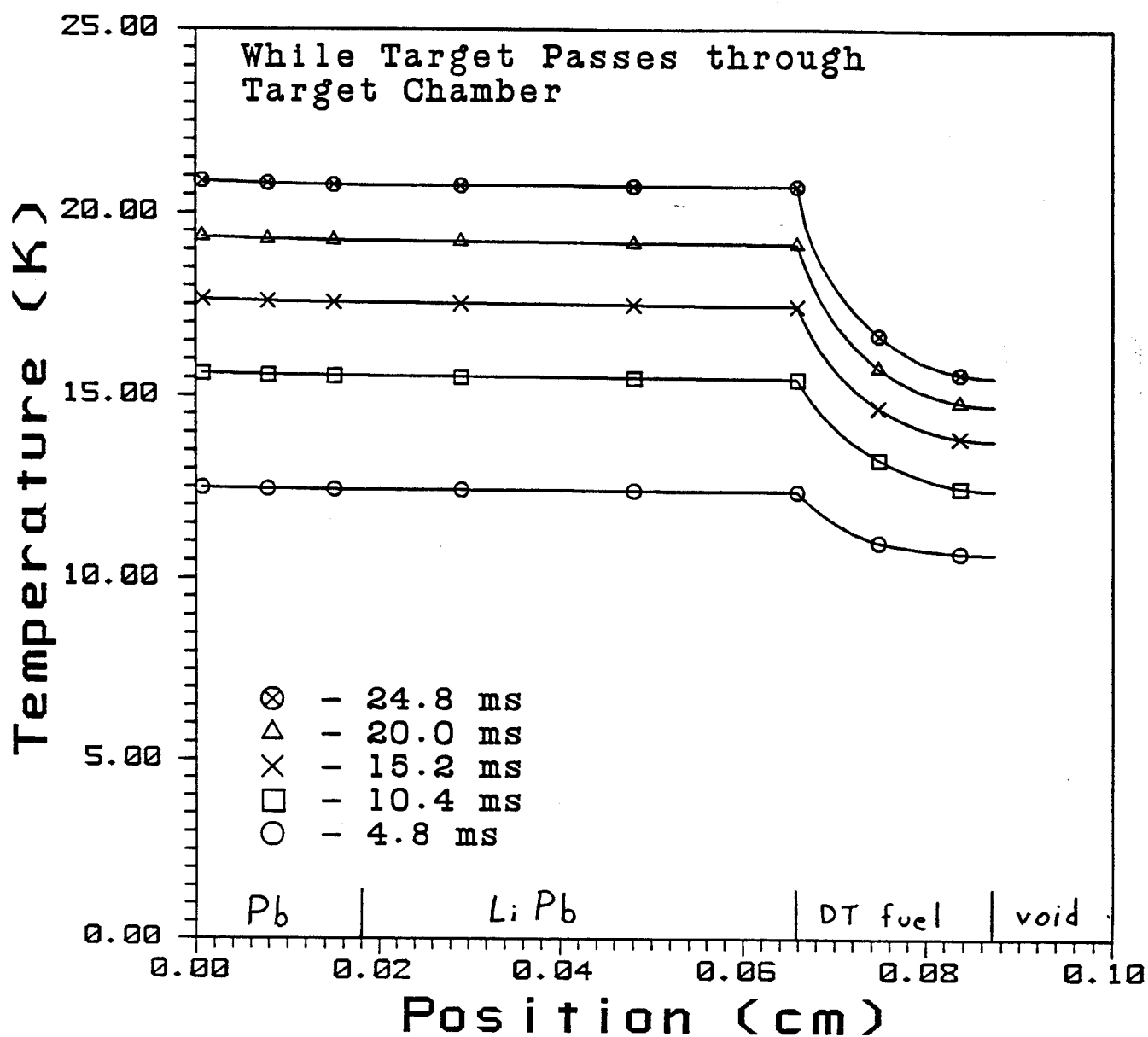
$$\begin{aligned} \ddot{q}_{\text{rad}} &= 5.7 \times 10^{-12} \text{ W/cm}^2 T_g^4 \\ &= 2.0 \text{ W/cm}^2 \end{aligned}$$

### **Total**

$$\ddot{q}_{\text{total}} = 5.5 \text{ W/cm}^2$$

## Target is Heated in Target Chamber

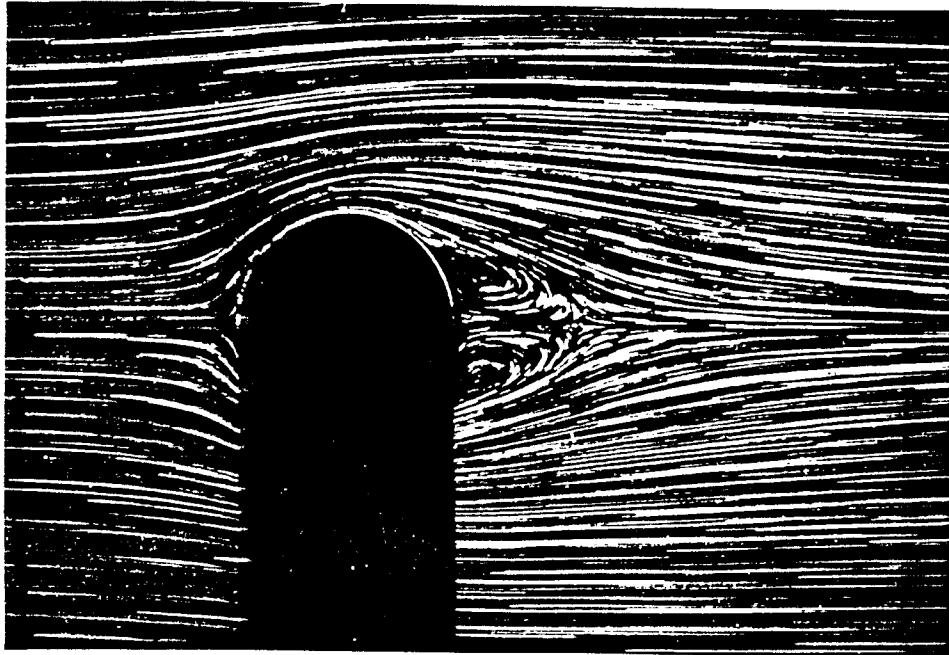
### TEMPERATURE IN CRYOGENIC TARGET



## *The Heat Transfer is Not Uniform Around the Target*

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Pattern for a low Re flow about a sphere (from An Album of Fluid Motion by Van Dyke)



51. Sphere at  $R=56.5$ . As in figure 8, the sphere is falling steadily down the axis of a tube filled with oil, but here so large that the influence of the walls is negligible. Magnesium

cuttings are illuminated by a sheet of light, which casts the shadow of the sphere. *Archives de l'Académie des Sciences de Paris. Payard & Coutanceau 1974*

Higher heat transfer at stagnation point on leading edge and at vortex ring at back of target.

We have no data for Reynolds numbers as low as 1000 but, from the trends at higher Re, we believe that the variation will be by no more than a factor of 2 from the average.

There may be some thermal smoothing of the temperature in the outer lead shell.

The effects of this nonuniform heat load on the fuel are unknown at this time.



## *Conclusions*

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- Sabot protects target during acceleration – target remains at 4 K until entrance to target chamber.
- Cryogenic fuel heats to slightly above 20 K – too hot for solid DT but acceptable for liquid DT in foam.
- Increasing injection velocity will improve situation, but only slowly because  $\ddot{q}_{\text{conv}} \propto v^6$ .

# *Cavity Calculations (CONRAD Code)*

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## PURPOSE

- 1) Calculate the radiation flux and pressure impulse at the INPORT tubes to find the mechanical stress on the tubes and the heating in the LiPb coating.
- 2) Determine the mass of LiPb vaporized from the INPORT tubes due to the prompt, high energy x-rays and the thermal radiation. This mass must be removed from the cavity gas, either by condensation or pumping, before the next target implosion.

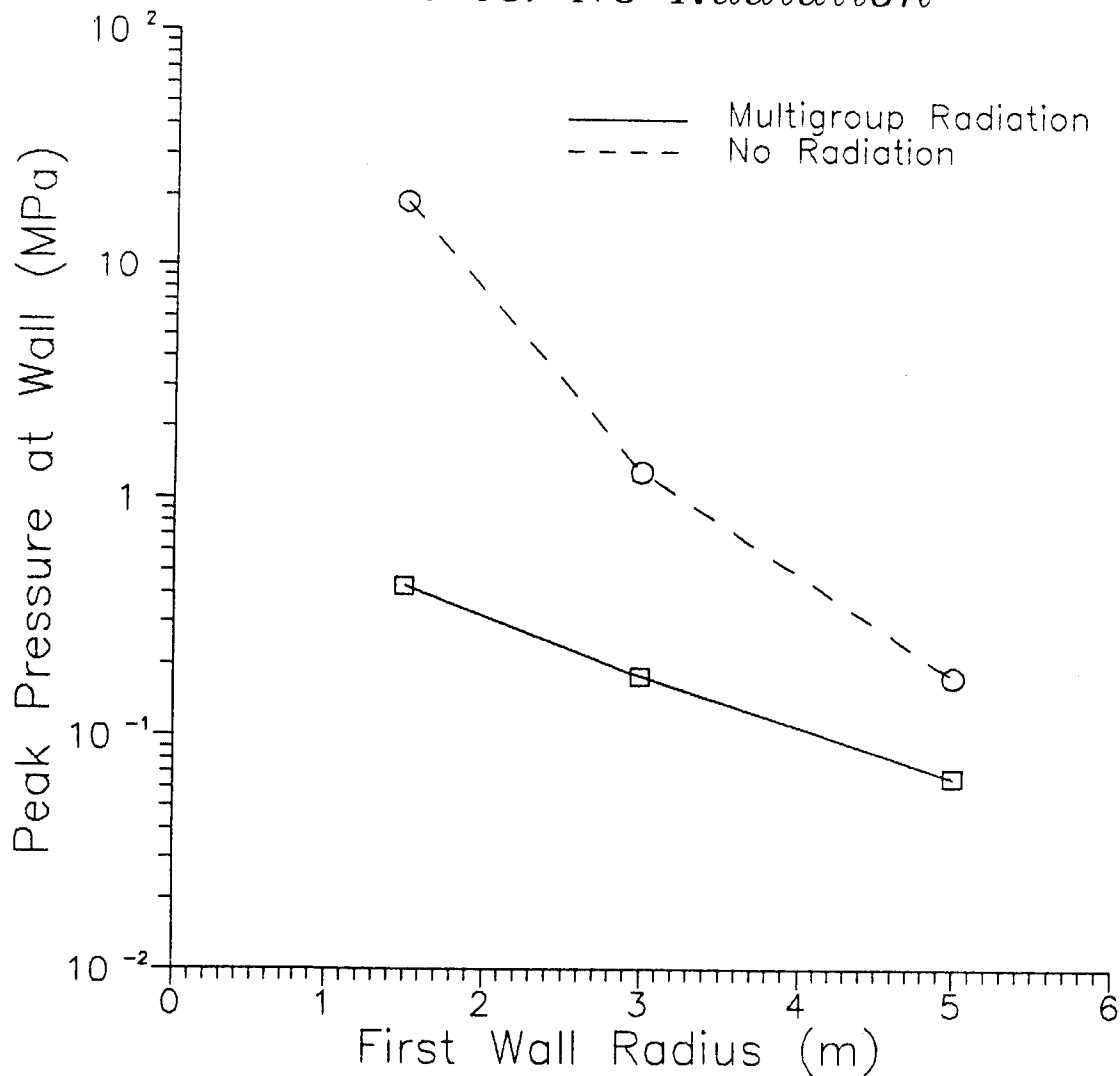
## INPUT FOR CALCULATIONS

- 1) Energy released from target:

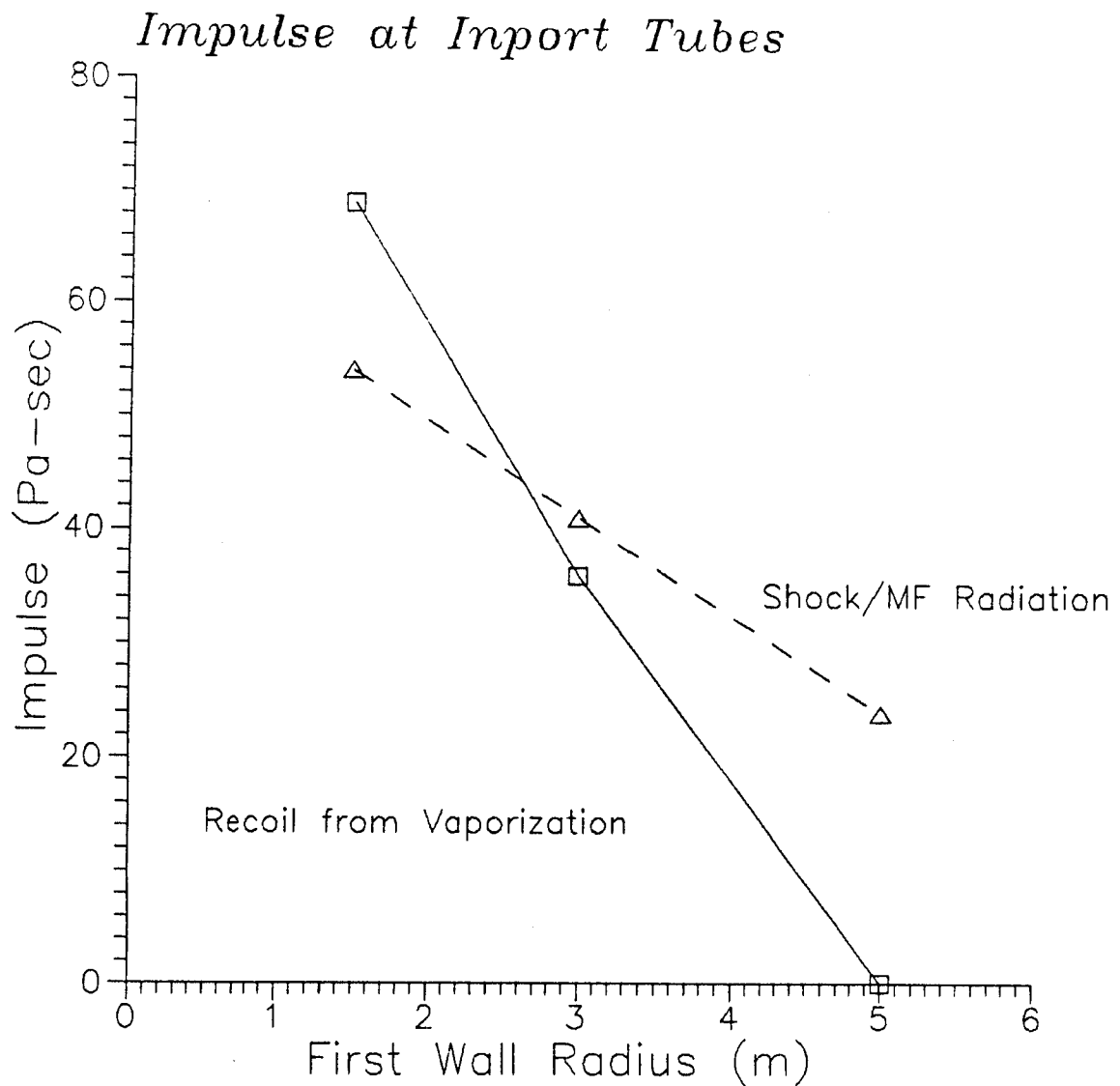
231 MJ	(neutrons)
63 MJ	(x-rays)
26 MJ	(ions)
<hr/>	
320 MJ	(total)

- 2) Background gas: argon, pressure = 10 torr
- 3) Distance to INPORT tubes = 3 m

*Peak Pressure at Inport Tubes --  
Radiation vs. No Radiation*

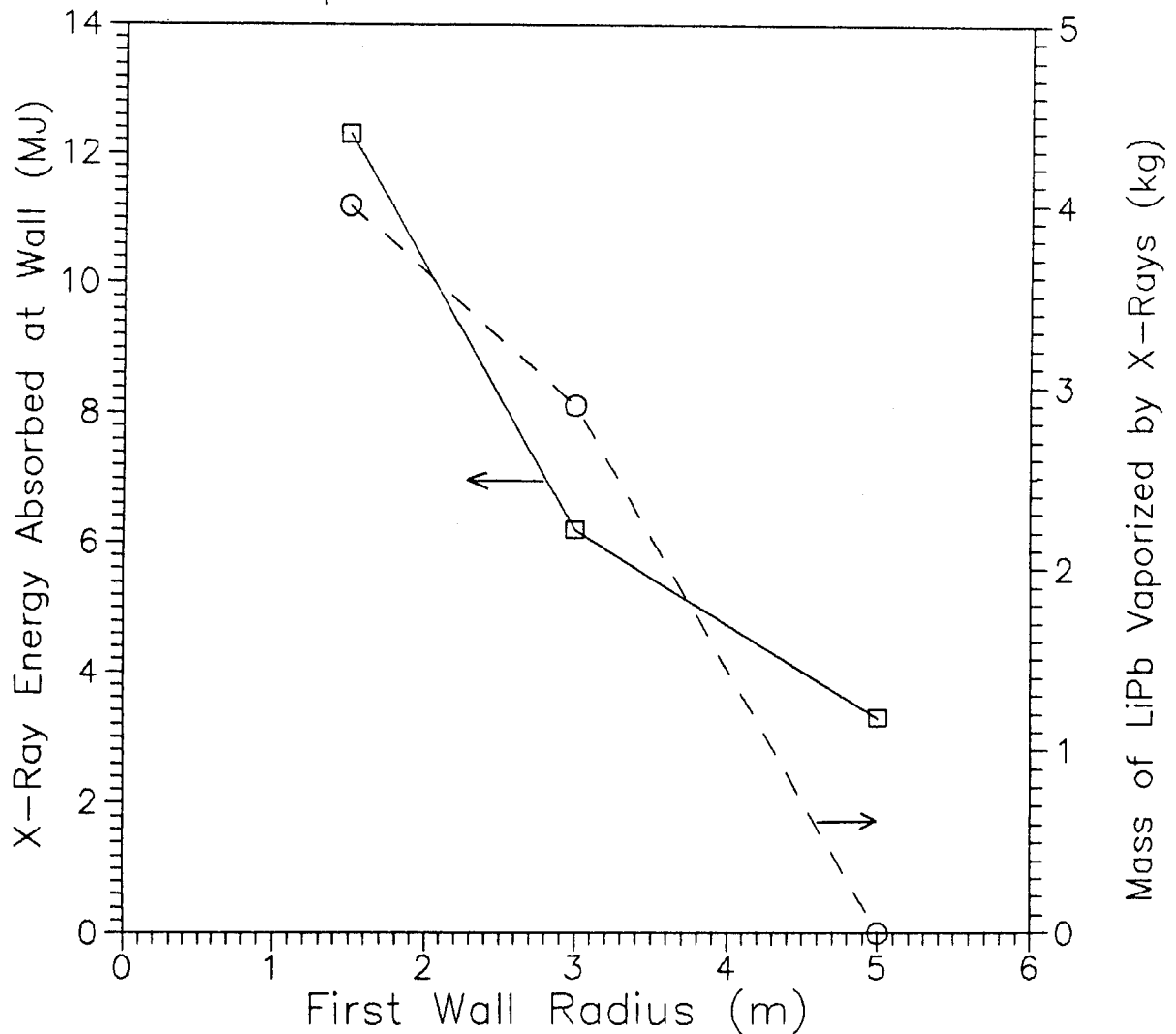


- Including radiation transport in the calculations significantly reduces the peak pressure of the shock front at the INPORT tubes.
- The pure hydrodynamics (no radiation) results provide an upper limit for the peak shock pressure.



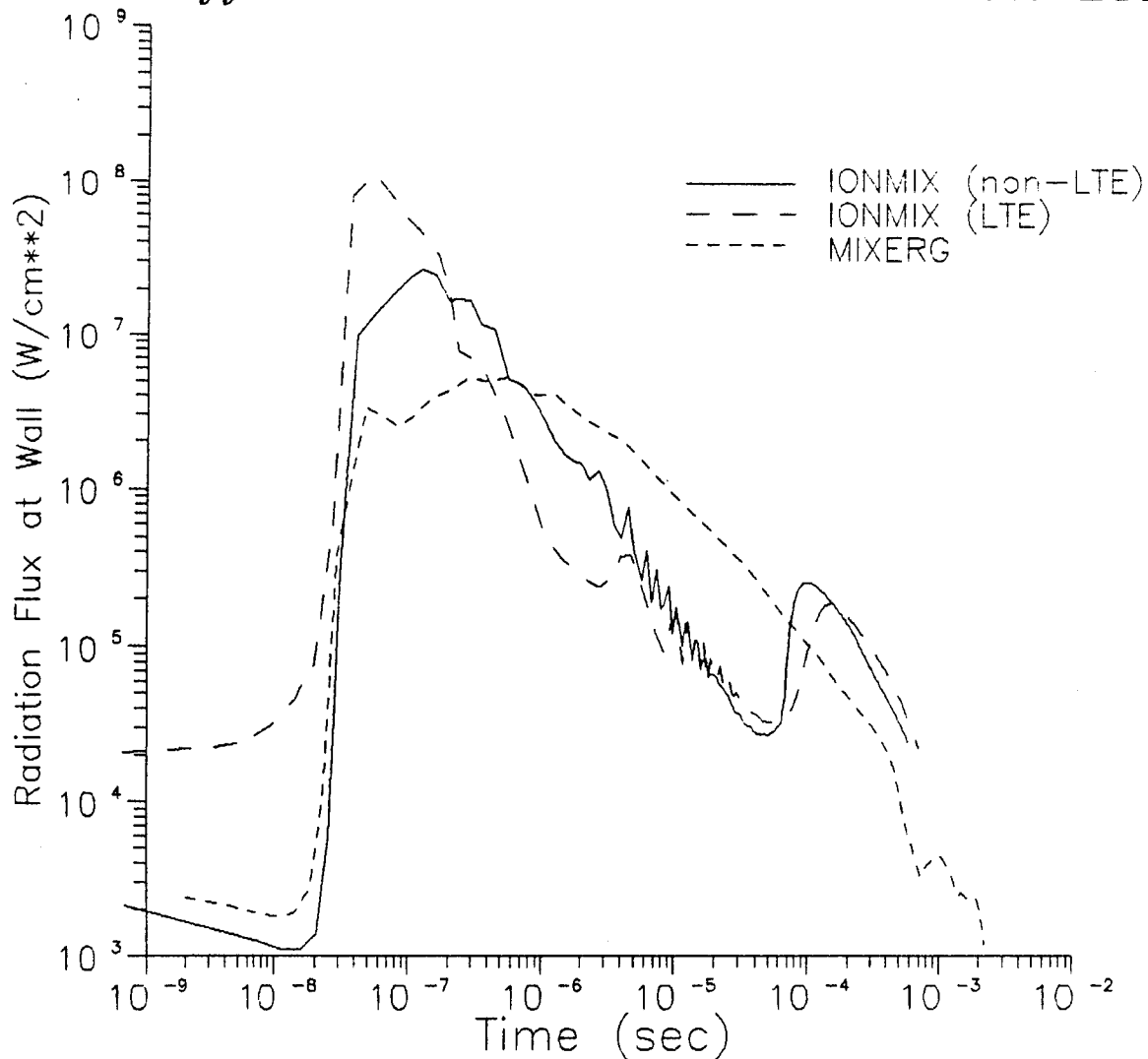
- The “recoil impulse” from the rapid vaporization of LiPb can exceed the shock-produced impulse when the distance to the tubes is  $\leq 2\frac{1}{2}$  meters.
- The total impulse on the INPORT tubes at 3 m is about 80 Pa-s.

# Prompt X-Rays Absorbed and Mass Vaporized vs. Wall Radius



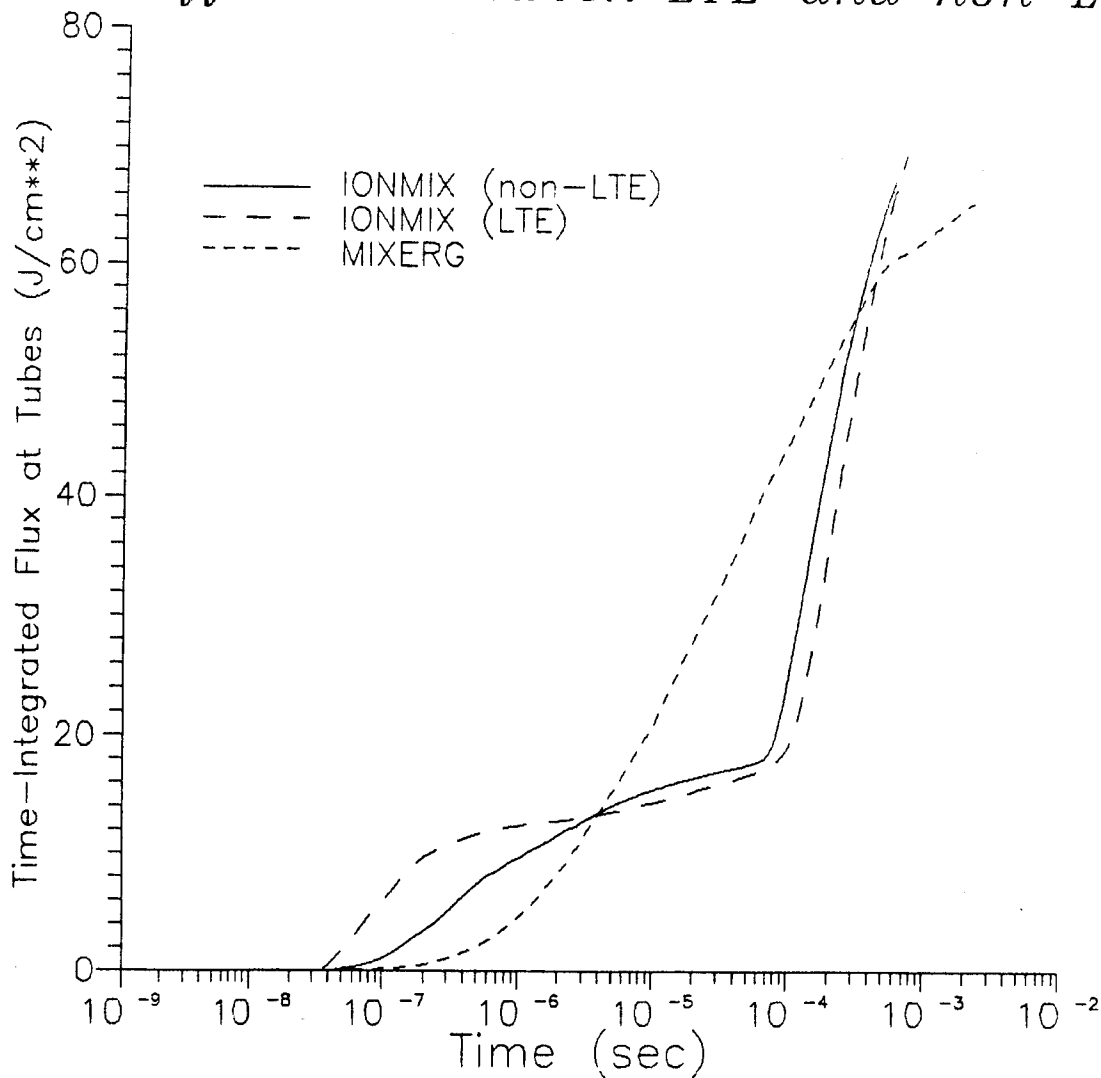
- At a distance of 3 m, roughly 90% of the target x-rays are absorbed by the background gas, so that  $\sim 6$  MJ are absorbed by the LiPb coating on the INPORT tubes.
- The mass of LiPb vaporized by the target x-rays is 2.9 kg. By comparison, the original cavity gas mass is 2.7 kg.

*Radiation Flux at Inport Tubes --  
Difference between LTE and non-LTE plasmas*



- We have developed a new equation of state and opacity code (IONMIX) that calculates the radiative properties of both LTE and non-LTE plasmas.
- The assumption of LTE leads to overestimating the radiative flux at the tubes at early times ( $\leq 0.2 \mu\text{s}$ ), and underestimates the flux at later times. (The solid line is expected to be the most accurate curve.)

*Time-Integrated Flux at Inport Tubes --  
Difference between LTE and non-LTE plasmas*



- Roughly 1/4 of the x-ray and ion energy is radiated to the INPORT tubes before the shock arrives at the tubes ( $t_{\text{SHOCK}} \approx 0.3$  ms).
- Non-LTE effects are noticeable when the background gas pressure is 10 torr, and become even more important at lower pressures.

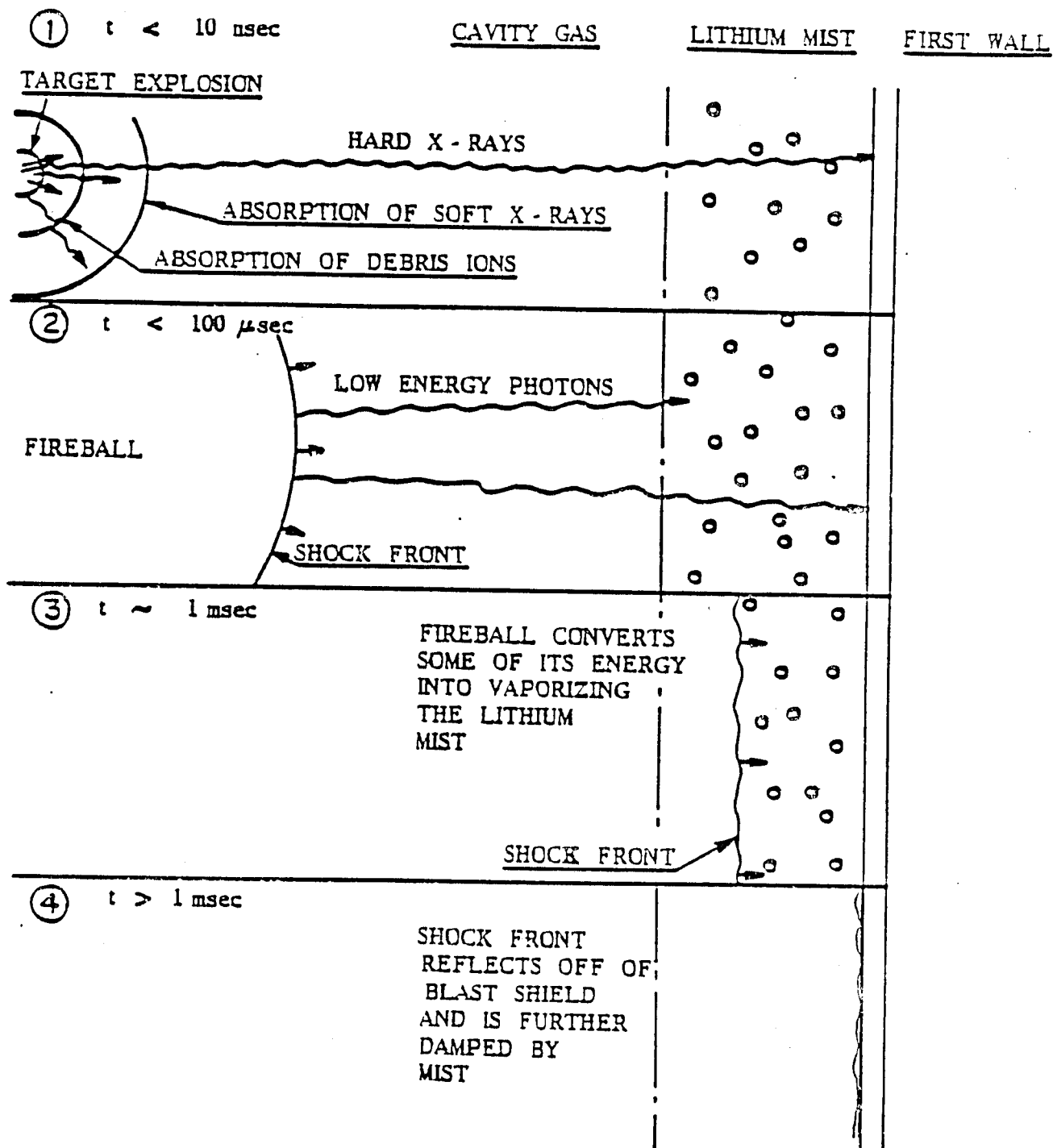
# *Features of the CONRAD Radiation-Hydrodynamics Code*

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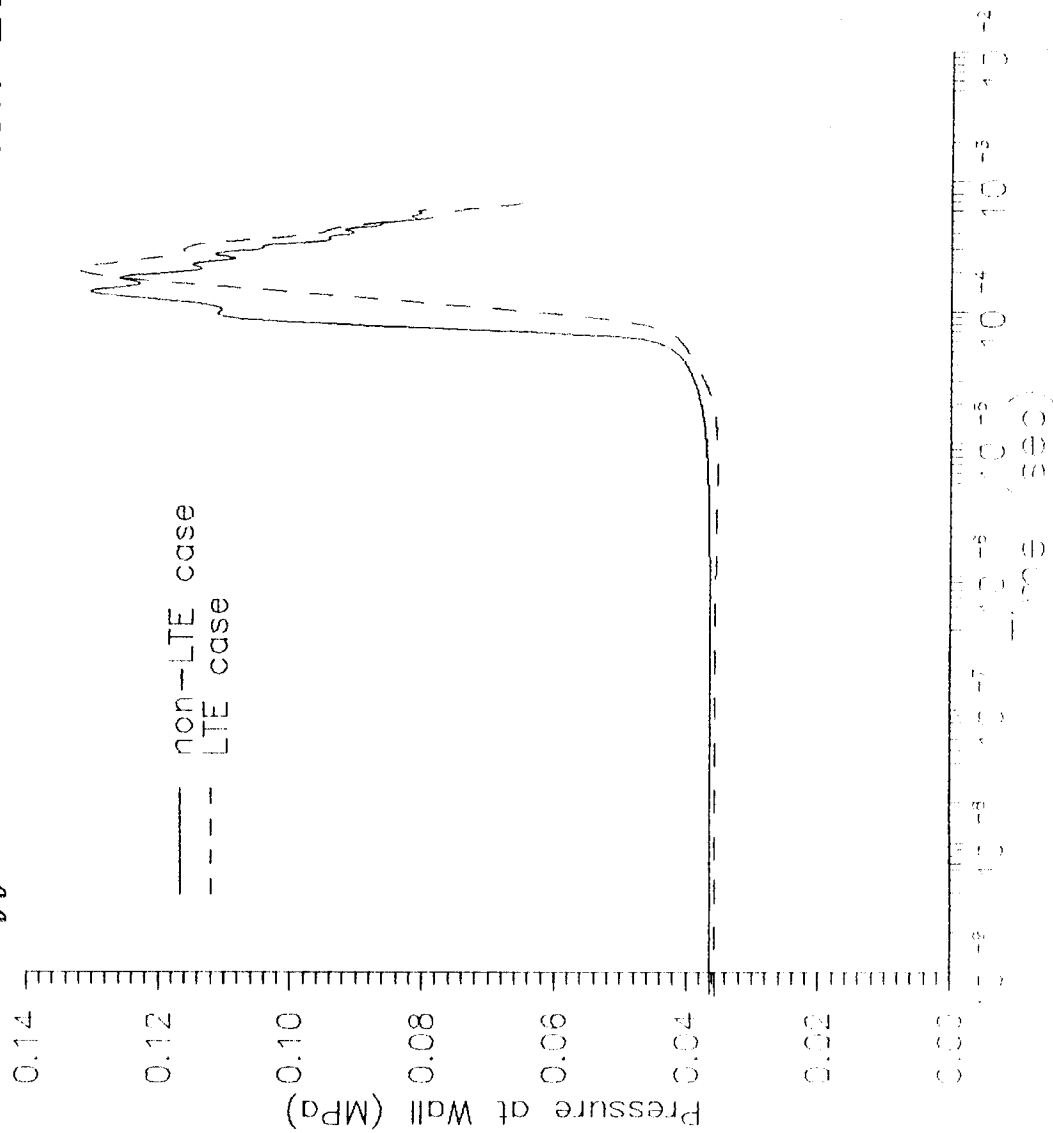
- One dimensional, one fluid lagrangian hydrodynamics with Von Neuman artificial viscosity
- One temperature plasma approximation ( $T_{\text{electron}} = T_{\text{ion}}$ )
- Multifrequency (20 group) flux limited diffusion of radiation
- Equation of state and opacities based on semi-classical atomic model (MIXERG code)
- Time dependent target point source x-ray attenuation in plasma using Bigg's cross section data
- Time dependent, energy dependent, species dependent, target ion stopping in plasma using combined bound electron, nuclear, and free electron stopping powers (SNL-Mehlhorn model)



# Physical Processes in ICF Target Chambers



*Pressure at Inport Tubes --  
Difference between LTE and non-LTE plasmas*



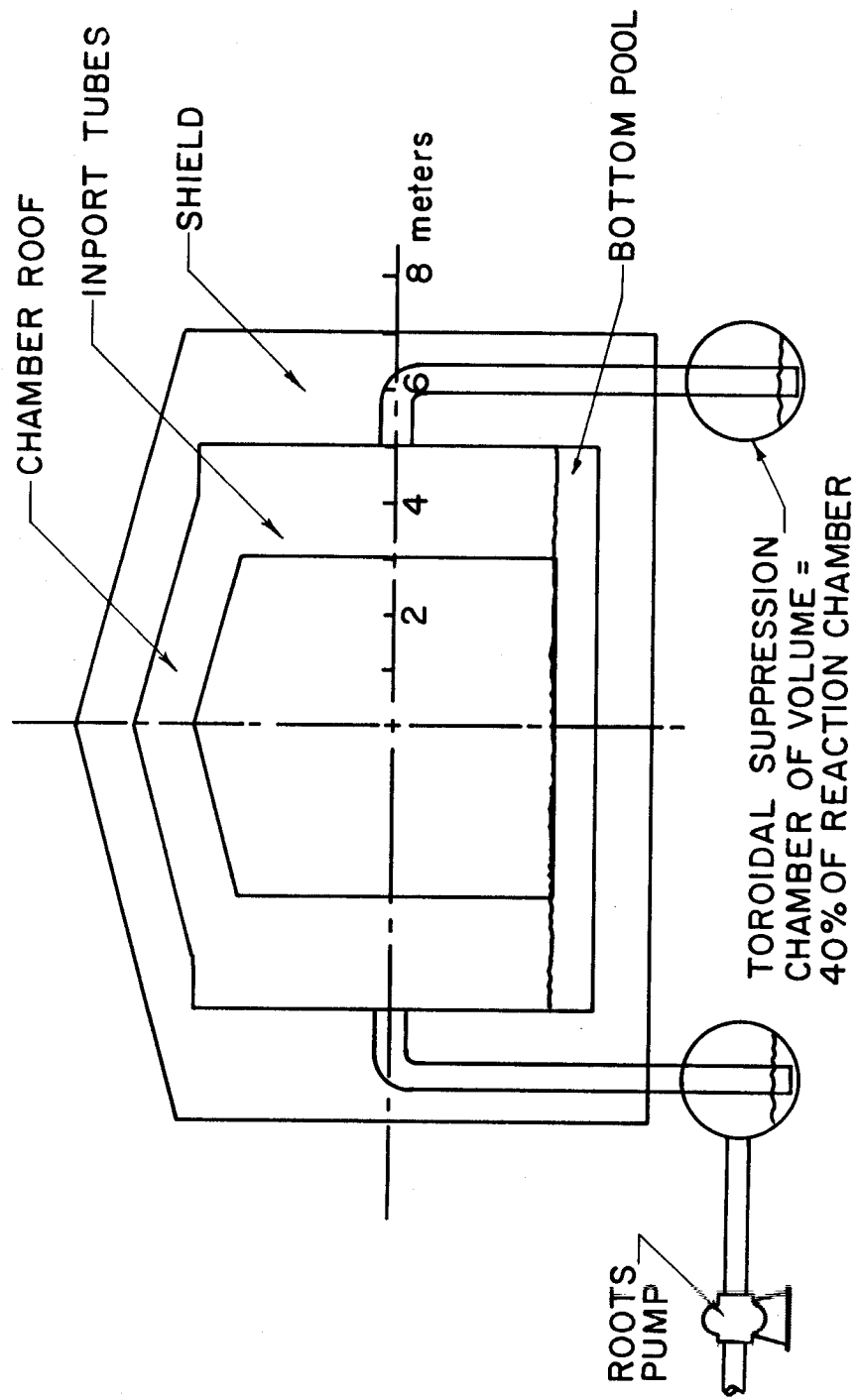
## General Parameters Used to Calculate Required Pumping Speed:

Radius to INPORT tubes (m)	3.0
Radius to Vacuum Wall (m)	5.0
Height from Bottom Pool to Roof (m)	6.0
Volume Fraction in INPORT Zone (%)	33
Gas in Chamber	Argon
Volume of Gas in INPORT Zone (m <sup>3</sup> )	202
Volume of Gas in Cavity Center (m <sup>3</sup> )	170
Pressure in Chamber Before Shot (torr)	26
Temperature in Chamber Before Shot (K)	800
Pressure in Chamber After Shot (torr)	290
Temperature in Chamber After Shot (K)	9000

## *Procedure and Assumptions Used to Calculate “LIBRA” Pumping Requirement*

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- It is assumed that a suppression chamber with LiPb cooling at 623 K is connected to the target chamber. The pressure in the suppression chamber is maintained at 26 torr.
- After the shot the gas in the target chamber expands isentropically into the suppression chamber and the pressure in both chambers equilibrates.
- Temperature of gas in the INPORT tube zone is assumed at 773 K (500°C) while that in the chamber center stays at the isentropic expansion temperature.
- As the temperature in the chamber cools down to 800 K the pressure falls below 26 torr and fresh argon is injected to replace that which was evacuated.
- The gas in the suppression chamber is pumped out until the pressure reaches 26 torr at 623 K.
- Pumping speed is calculated while varying the volume of the suppression chamber and using a rep rate of 3 Hz.

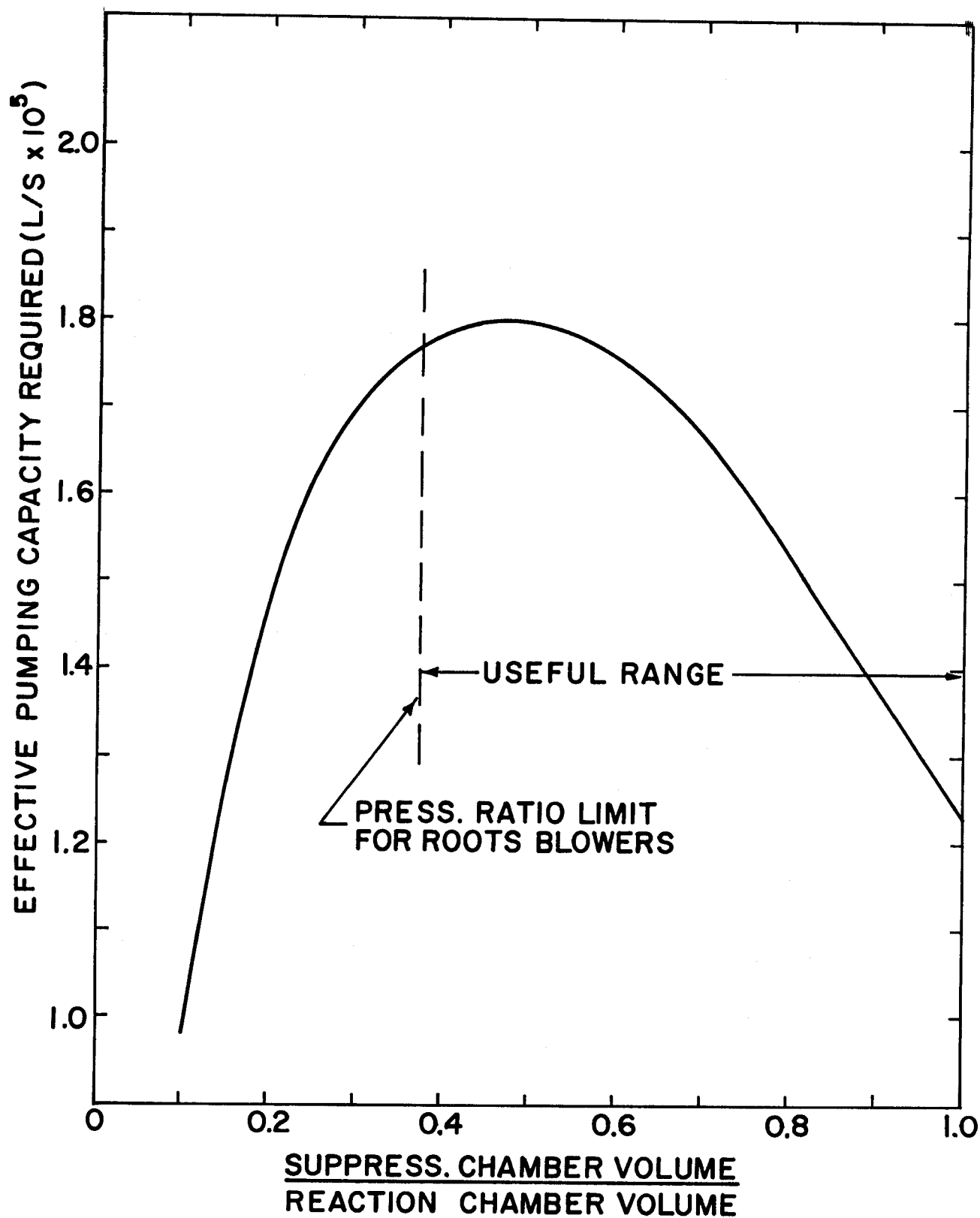


**SCHEMATIC CROSS SECTION OF LIBRA CHAMBER  
WITH A SUPPRESSION CHAMBER SHOWN TO SCALE**

$$\text{FOR } V_S/V_T = 0.4$$

# *Pumping Parameters for LIBRA*

Ratio of Sup- pression Cham- ber Volume to Target Cham- ber Volume	Gas Tempera- ture In Target Chamber After Isentropic Ex- pansion (K)	Gas Average Tem- perature in Tar- get Chamber Af- ter Equilibrium with Suppres- sion Chamber (K)	Gas Average Pres- sure in Target Chamber and Suppression Cham- ber After Equi- librium (torr)	Ratio of Pres- sure After Equi- librium to Steady State Pressure	Fraction of Tar- get Chamber Gas Mass Exhausted per Shot (%)	Pumping Speed Required (l/s)
1.0	1440	1076	29	1.12	16.4	$1.23 \times 10^5$
0.8	1652	1173	30.8	1.47	19.3	$1.53 \times 10^5$
0.5	2193	1420	35.2	1.76	23.4	$1.80 \times 10^5$
0.4	2496	1557	38.2	1.95	24.4	$1.78 \times 10^5$
0.3	2955	1767	42.9	2.21	25.3	$1.70 \times 10^5$
0.2	3628	2052	50.3	2.57	24.4	$1.47 \times 10^5$
0.1	4950	2647	68	2.60	21	$1.08 \times 10^5$



**EFFECTIVE PUMPING REQUIREMENT FOR LIBRA  
AS A FUNCTION OF SUPPRESSION CHAMBER VOLUME**

## *Results and Conclusions*

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- A pressure ratio of  $\geq 2$  is not recommended for roots blowers for sudden pressure fluctuations as is the case in LIBRA. This limits the operation to a range of suppression chamber to target chamber volume ratio of  $\geq 0.4$ .
- As the suppression chamber/target chamber volume ratio decreases, the pumping requirement increases, peaking at  $V_S/V_T=0.5$ .
- The most cost effective system trades the price of the suppression chamber with the cost of increased pumping.
- An effective pumping speed of  $< 2 \times 10^5 \text{ } \ell/\text{s}$  and  $> 1.2 \times 10^5 \text{ } \ell/\text{s}$  is required for the LIBRA target chamber, depending on the size of the suppression chamber.





# **STATUS OF LIBRA NEUTRONICS**

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**Mohamed Sawan**

**Fusion Power Associates**

**28–29 January 1988**

LIBRA neutronics activities carried out during the last year (1987) are summarized below.

1) Neutronics benchmark calculations

One-dimensional calculations have been performed for different blanket designs by FPA (using ONEDANT) and KfK (using MCNP) and results compared.

2) Target neutronics calculations

Neutronics calculations have been performed for the 3.2 mg LIBRA target to determine the energy spectra of emitted neutrons and gamma photons as well as the breakdown of neutron, gamma, x-rays, and debris yields.

3) Chamber neutronics analysis

One-dimensional neutronics analysis has been performed for the LIBRA chamber to determine design options that satisfy the tritium breeding and wall protection requirements.

## ***Neutronics Information Sent to KfK***

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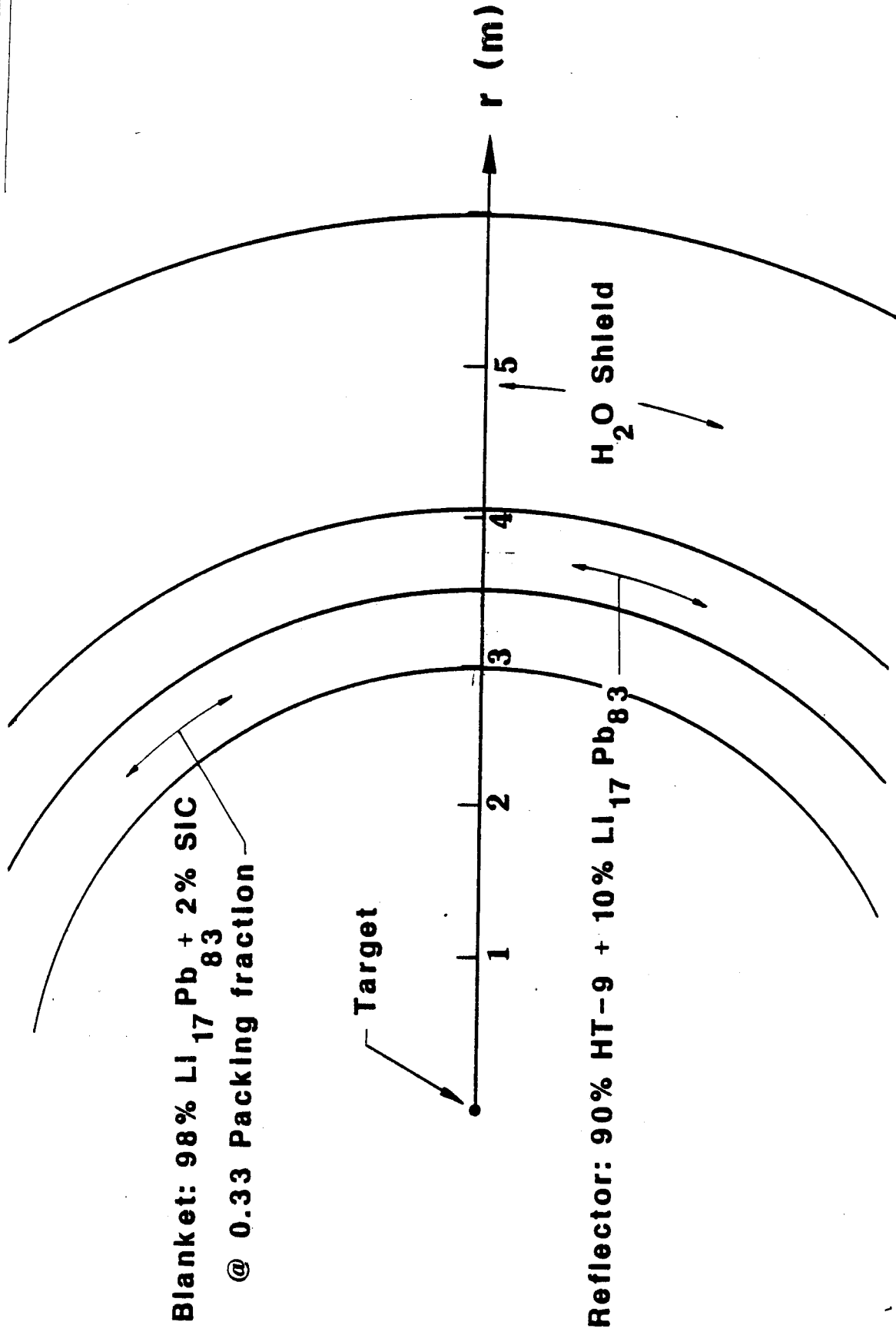
<b>Date</b>	<b>Item</b>
<b>2 July 1987</b>	<b>Nuclide densities in blanket and reflector</b>
<b>15 July 1987</b>	<b>Target spectrum used for benchmark calculations</b>
<b>20 July–26 Nov. 1987</b>	<b>10 communications related to benchmark</b>
<b>2 October 1987</b>	<b>Target geometry, composition, and results of target neutronics</b>

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## *Neutronics Benchmark Calculations*

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- Neutronics calculations performed by FPA using the ONEDANT code,  $P_3$ - $S_8$  approximation, spherical geometry, 30 n-12  $\gamma$  group cross section data based on ENDF/B-V, and the HIBALL target spectrum. Ten different cases were analyzed using different blanket thicknesses ( $\Delta_B$ ), different reflector thicknesses ( $\Delta_r$ ) and different lithium enrichments ( $\%{}^6\text{Li}$ ).



**SCHEMATIC OF NEUTRONICS CALCULATIONAL MODEL**

# Nuclear Parameters Calculated by FPA for the 10 Cases

Case	$\Delta_B$ (m)	$\Delta_r$ (m)	% <sup>6</sup> Li	TBR	$M_n$	$M^*_o$	Peak HT-9
							dpa/FPY
1	2	.5	7.42	1.152	1.366	1.250	2.44
2	2	.5	90	1.625	1.259	1.174	1.21
3	1.5	.5	7.42	0.957	1.407	1.280	5.37
4	1.5	.5	90	1.659	1.278	1.189	3.12
5	1.0	.5	7.42	0.715	1.451	1.312	11.63
6	1.0	.5	90	1.406	1.293	1.200	7.95
7	1.0	.5	35	1.168	1.346	1.237	9.89
8	0.55	.5	90	1.162	1.309	1.211	18.00
9	1.0	.8	35	1.175	1.377	1.258	9.71
10	0.55	0.85	90	1.177	1.374	1.238	17.61

$$M_o = \frac{(0.28 + 0.72M_n)0.99E_{DT}}{E_{DT}}$$

- Tritium breeding ratio calculations have been performed by KfK for cases 2, 4, 6, 8, and 10 using the MCNP version 3 code and different MCNP continuous energy cross section libraries available to KfK. The geometrical model, nuclide densities, target spectrum used are identical to those used by FPA. Analog Monte Carlo was used with no variance reduction method. The E-mail service was used to efficiently compare results and exchange comments between FPA and KfK.



## Comparison Between TBR Results Obtained by FPA and KfK

Case	FPA ONEDANT Results	KfK MCNP Results		
		23 July 87	1 Nov. 87	12 Nov. 87
2	1.625	1.659	1.670	—
4	1.549	1.580	1.591	1.570
6	1.406	1.427	1.442	1.409
8	1.162	1.159	1.193	1.138
10	1.177	1.245	1.254	1.173
Comments	Used as reference for comparison.	Results agree to within 2% except for case 10 the difference is 5.5%. This appears to be due to an overesti- mate of T <sub>2</sub> produc- tion in the reflector (27% higher than case 8).	Used ENDF/B-IV ex- cept for <sup>6</sup> Li, <sup>12</sup> C, and Si. Re- sults agree to within 2.7% except for case 10 the difference is 6.5%.	Used ENDL85 ex- cept for <sup>6</sup> Li and <sup>7</sup> Li.  All results agree to within 2%.

## *Conclusions*

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- Differences between TBR values calculated by FPA and KfK are attributed to:
  - Different calculational methods (discrete ordinates vs. Monte Carlo)
  - Different energy treatments for cross sections (multigroup vs. continuous energy)
  - Different cross section data (ENDF/B-V vs. ENDF/B-IV or ENDL85)
- Excellent agreement ( $<2\%$  difference) is obtained between the ONEDANT results using ENDF/B-V data and the MCNP results using the most recent ENDL85 data.

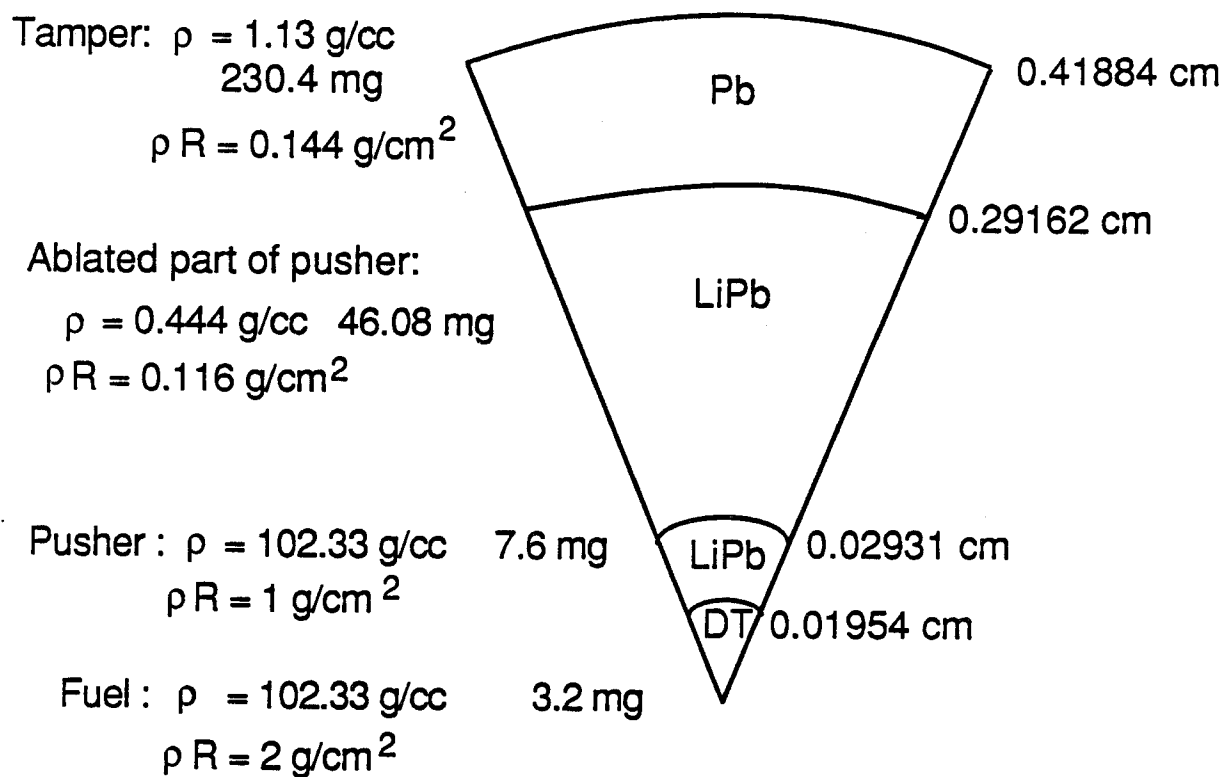
# ***Target Neutronics Calculations***

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**CODE: ONEDANT**

**Data: 30 n – 12  $\gamma$  group structure  
ENDF/B-V data**

**Source: Uniform 14.1 MeV neutron source in compressed  
DT fuel zone**



### LIBRA Target Configuration at Ignition

## ***Nuclear Energy Deposition in Target***

<b>Region</b>	<b>Neutron Energy Deposition (MeV/DT fusion)</b>	<b>Gamma Energy Deposition (MeV/DT fusion)</b>
<b>1</b>	<b>1.1966</b>	<b><math>1.74 \times 10^{-4}</math></b>
<b>2</b>	<b>0.1423</b>	<b><math>1.27 \times 10^{-3}</math></b>
<b>3</b>	<b>0.0352</b>	<b><math>1.73 \times 10^{-4}</math></b>
<b>4</b>	<b>0.0012</b>	<b><math>2.26 \times 10^{-4}</math></b>
<b>Total</b>	<b>1.3753</b>	<b><math>1.84 \times 10^{-3}</math></b>

**Total energy deposited by neutrons and gamma photons  
in target = 1.377 MeV/DT fusion**

## ***Spectra of Neutrons and Gamma Photons Emitted from the Target***

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**Energy carried by neutrons  
emitted from target**                      **12.418 MeV/DT fusion**

**Number of neutrons emitted  
from target (70.65% of  
neutrons at 14.1 MeV)**                      **1.0285 n/DT fusion**

**Average energy of neutrons  
emitted from target**                      **12.07 MeV**

**Energy carried by gamma  
photons emitted from target**                      **0.24 MeV/DT fusion**

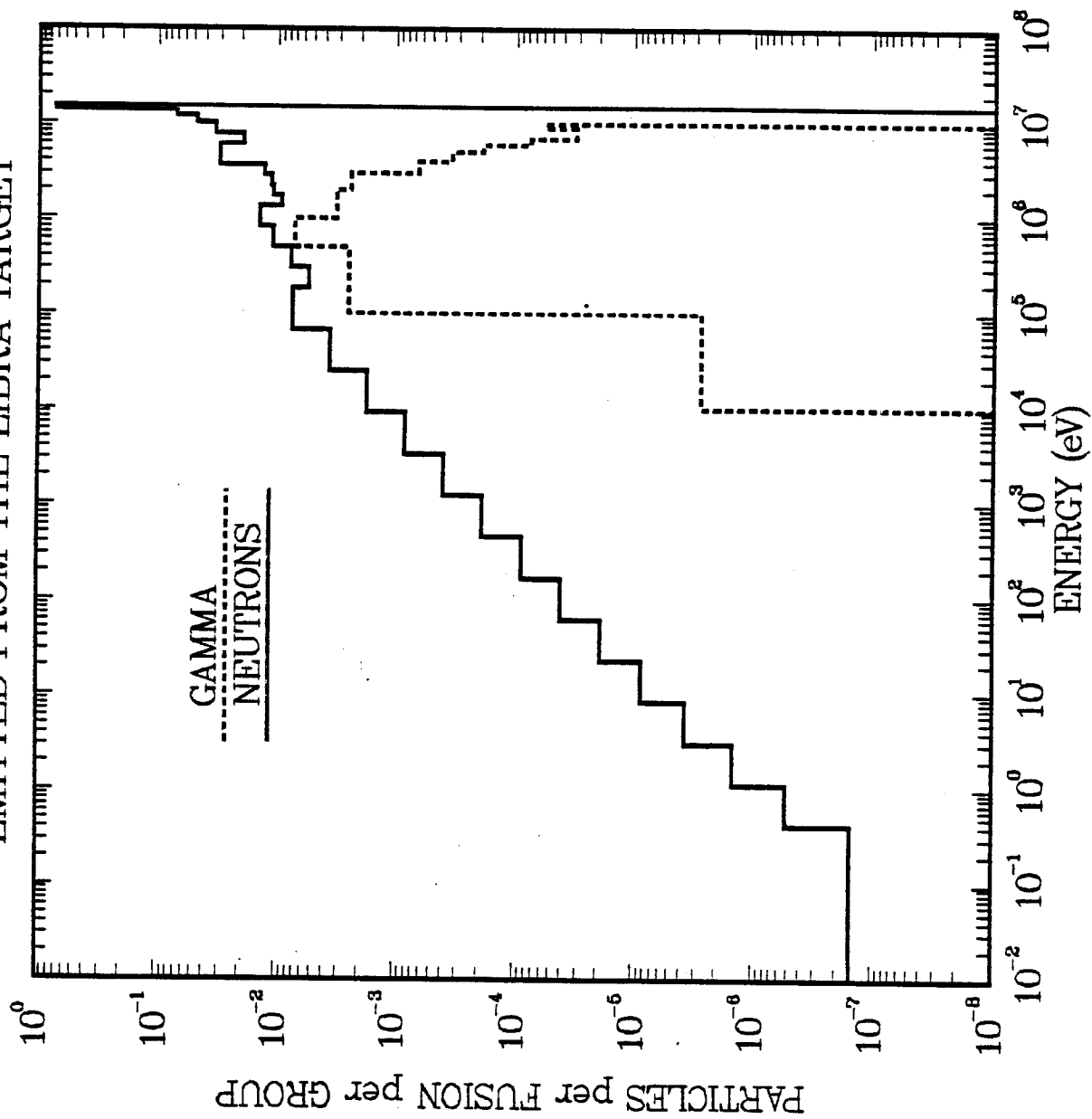
**Number of gamma photons  
emitted from target**                      **0.0168  $\gamma$ /DT fusion**

**Average energy of gamma  
emitted from target**                      **1.4 MeV**

**Energy lost in endoergic reactions =**

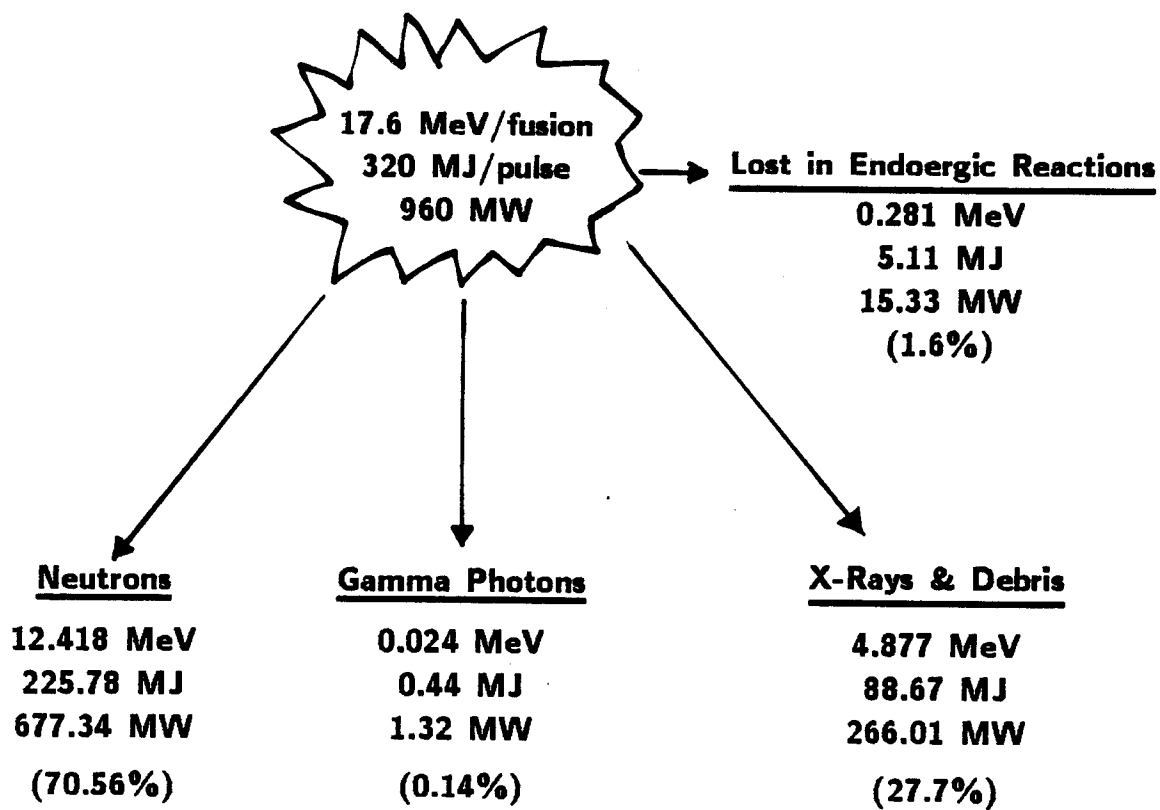
$$\begin{aligned} &14.1 - 12.418 - 0.024 - 1.377 \\ &= 0.281 \text{ MeV/DT fusion} \end{aligned}$$

ENERGY SPECTRA OF NEUTRONS AND GAMMA PHOTONS  
EMITTED FROM THE LIBRA TARGET



# Energy Flow for LIBRA Target

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## ***Chamber Neutronics Analysis***

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- ONEDANT spherical geometry calculations
- 30 n - 12  $\gamma$  group cross section data based on ENDF/B-V
- Point source at center of chamber emitting neutrons and gamma photons having the spectra calculated for the LIBRA target
- Normalized to 320 MJ DT yield and 3 Hz repetition rate
- Cavity radius = 3 m
- INPORT tubes with 0.33 packing fraction  
2% SiC and 98%  $\text{Li}_{17}\text{Pb}_{83}$  in tubes
- 0.5 m thick reflector consisting of 90% HT-9 and 10%  $\text{Li}_{17}\text{Pb}_{83}$

## ***Design Objectives***

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- **TBR  $\geq 1.1$**
- **Peak end-of-life dpa in HT-9  $\leq 200$  dpa. For 30 FPY reactor life the peak dpa rate should not exceed 6.6 dpa/FPY**
- **Maximize energy multiplication**
- **Minimize blanket thickness**

### **Design Parameters Varied**

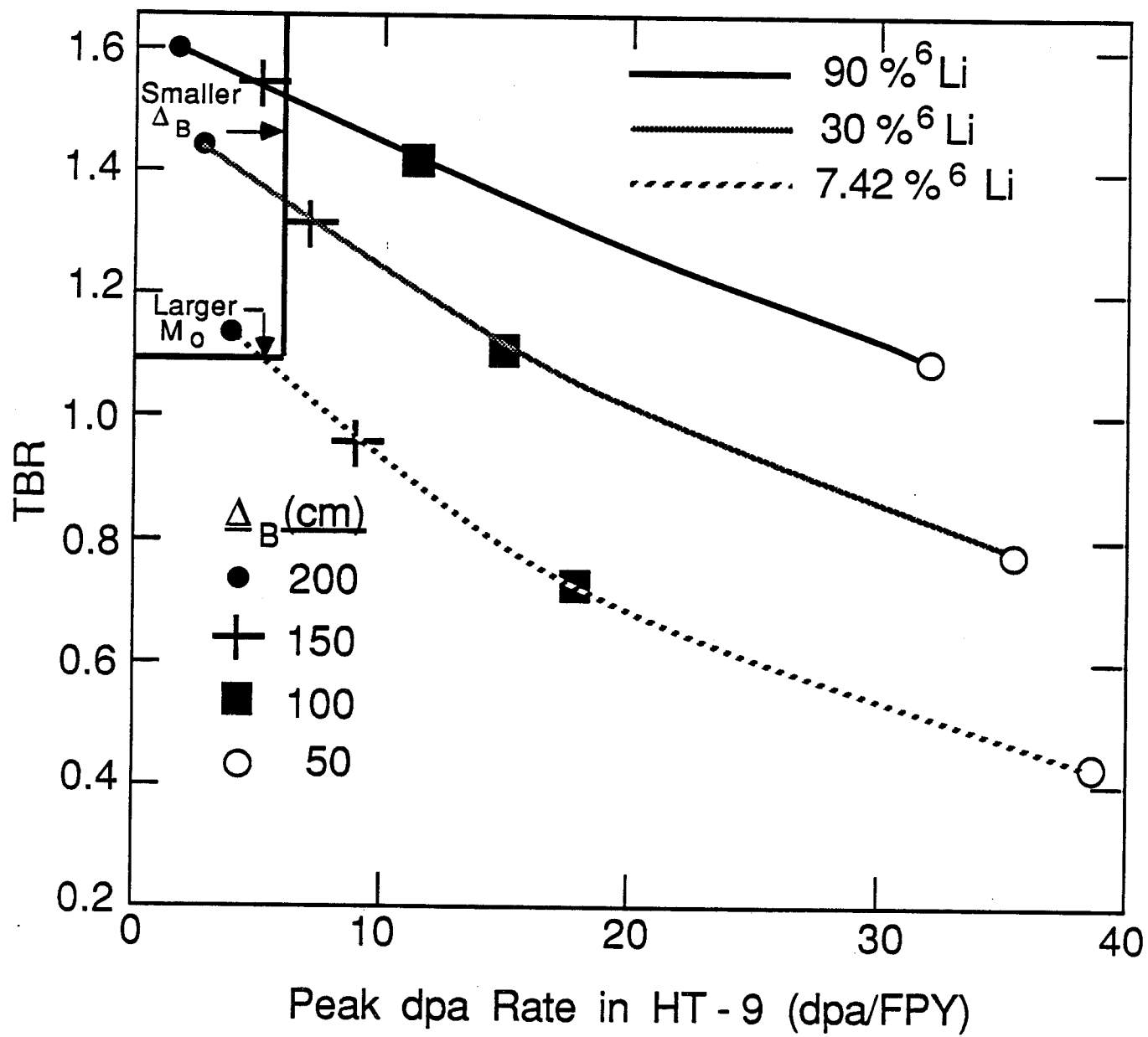
Blanket thickness	$\Delta_B$
Lithium enrichment	% <sup>6</sup> Li

Case	% <sup>6</sup> Li	$\Delta B$ (cm)	TBR	$M_n^*$	$M_o^{**}$	Peak dpa/FPY in HT-9	Peak He appm/FPY in HT-9
1	7.42	50	0.421	1.465	1.313	38.87	75.46
2	7.42	100	0.705	1.450	1.302	18.40	12.91
3	7.42	150	0.944	1.408	1.273	8.51	2.16
4	7.42	200	1.136	1.366	1.243	3.88	0.37
5	30	50	0.791	1.381	1.253	36.00	75.39
6	30	100	1.108	1.357	1.236	16.07	12.89
7	30	150	1.312	1.323	1.212	6.99	2.15
8	30	200	1.445	1.296	1.193	3.00	0.36
9	90	50	1.108	1.311	1.204	31.16	75.18
10	90	100	1.387	1.294	1.192	12.60	13.57
11	90	150	1.528	1.274	1.178	4.95	2.14
12	90	200	1.603	1.260	1.168	1.92	0.36

$$*M_n = \text{Nuclear energy multiplication} = \frac{\text{Energy deposited by } n \text{ and } \gamma \text{ in blanket and reflector}}{\text{Energy of } n \text{ and } \gamma \text{ incident on blanket}}$$

$$**M_o = \text{Overall energy multiplication} = \frac{\text{Energy deposited by } n, \gamma, X, \text{ and } D \text{ in blanket and refl.}}{\text{DT yield}}$$

$$M_o = 0.984 (0.7185M_n + 0.2815)$$



**LIBRA Blanket Design Options Satisfying TBR and  
Damage Requirements (no reflector replacement)**

<b>% <math>^6\text{Li}</math></b>	<b><math>\Delta_B</math> (cm)</b>	<b>TBR</b>	<b><math>M_o</math></b>	<b>dpa/FPY</b>	<b>He appm/FPY</b>
90	135	1.5	1.183	6.6	3.5
30	153	1.32	1.215	6.6	2.0
7.42	190	1.1	1.248	4.4	0.8

**LIBRA Blanket Design Options Satisfying TBR  
Requirement with Replaceable Reflector**

<b>% <math>^6\text{Li}</math></b>	<b><math>\Delta_B</math> (cm)</b>	<b>TBR</b>	<b><math>M_o</math></b>	<b>dpa/FPY</b>	<b>He appm/FPY</b>	<b>Number of Replacements</b>
90	50	1.1	1.204	31.2	75.2	4
30	100	1.1	1.236	16.07	12.9	2

## *Conclusions*

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- If the reflector is designed as a lifetime component, the thinnest blanket can be designed with 90%  $^6\text{Li}$  leading to a minimum channel length of 4.85 m. The energy multiplication is relatively low and excessive  $T_2$  breeding is obtained.
- If reflector replacement is allowed, using 90%  $^6\text{Li}$  yields the thinnest blanket with a channel length as small as 4 m being possible. The reflector has to be replaced four times during the reactor life.
- Using natural Li in the LiPb blanket yields the largest energy multiplication but the channel length should be at least 5.4 m.



# ***INPORT Response Code Development***

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- **Planar – linear motion**

- Fundamental mode only**

- **Planar – linear motion**

- Fifteen modes**

- **Planar – linear motion**

- Fifteen modes**

- Coriolis acceleration (fluid)**

- **Planar – nonlinear motion**

- Fifteen modes**

- Coriolis acceleration (fluid)**

- Cubic transverse displacement terms**

- **Nonplanar motion**

- Fifteen modes**

- Coriolis acceleration (fluid)**

- Cubic transverse and lateral displacement terms**



**VIBRATION AND STABILITY  
OF VERTICAL TUBES CONVEYING FLUID  
SUBJECTED TO PLANAR EXCITATION**

by

**Roxann L. Engelstad**

**A thesis submitted in partial fulfillment of  
the requirements for the degree of**

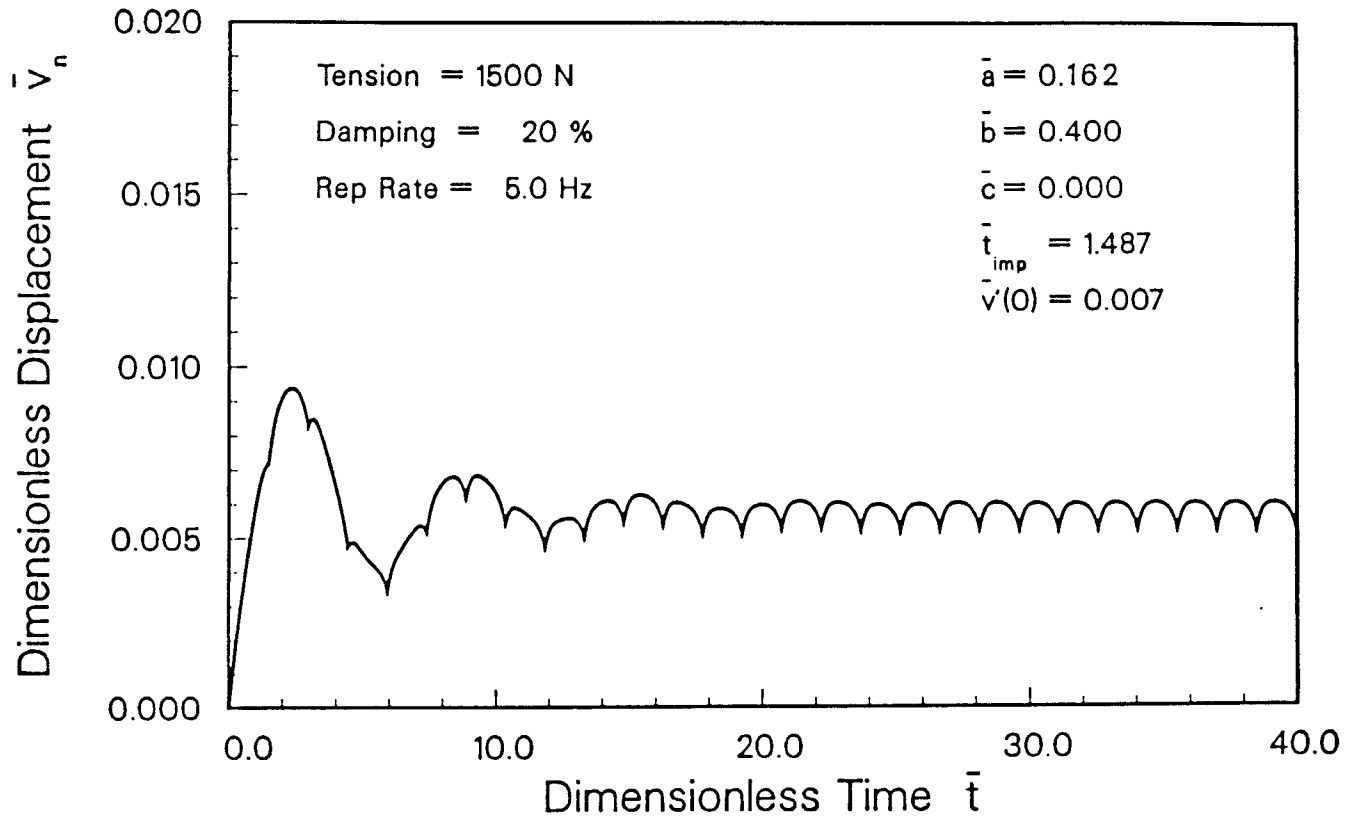
**DOCTOR OF PHILOSOPHY  
(Engineering Mechanics)**

at the

**UNIVERSITY OF WISCONSIN-MADISON**

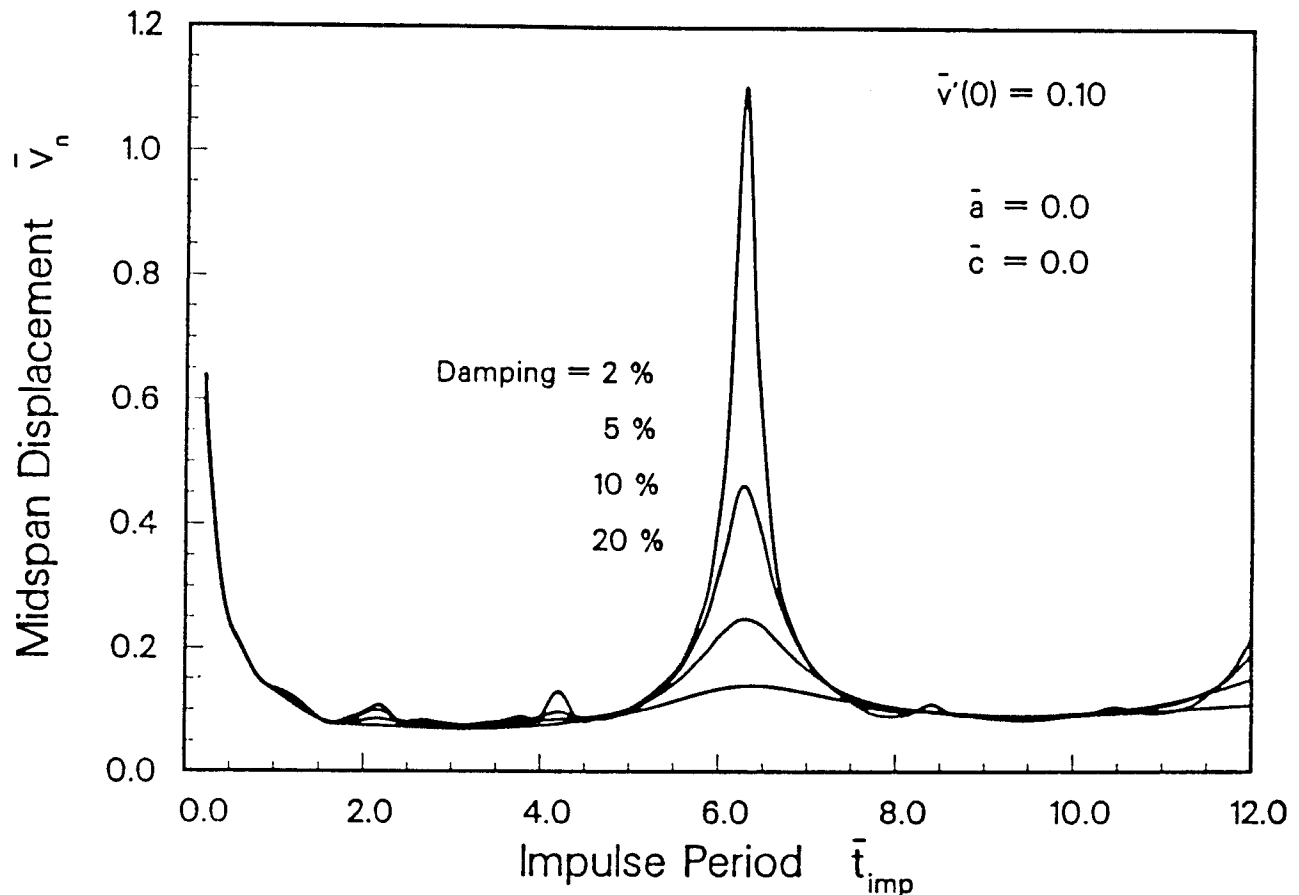
**1988**

## INPORT Midspan Displacement History



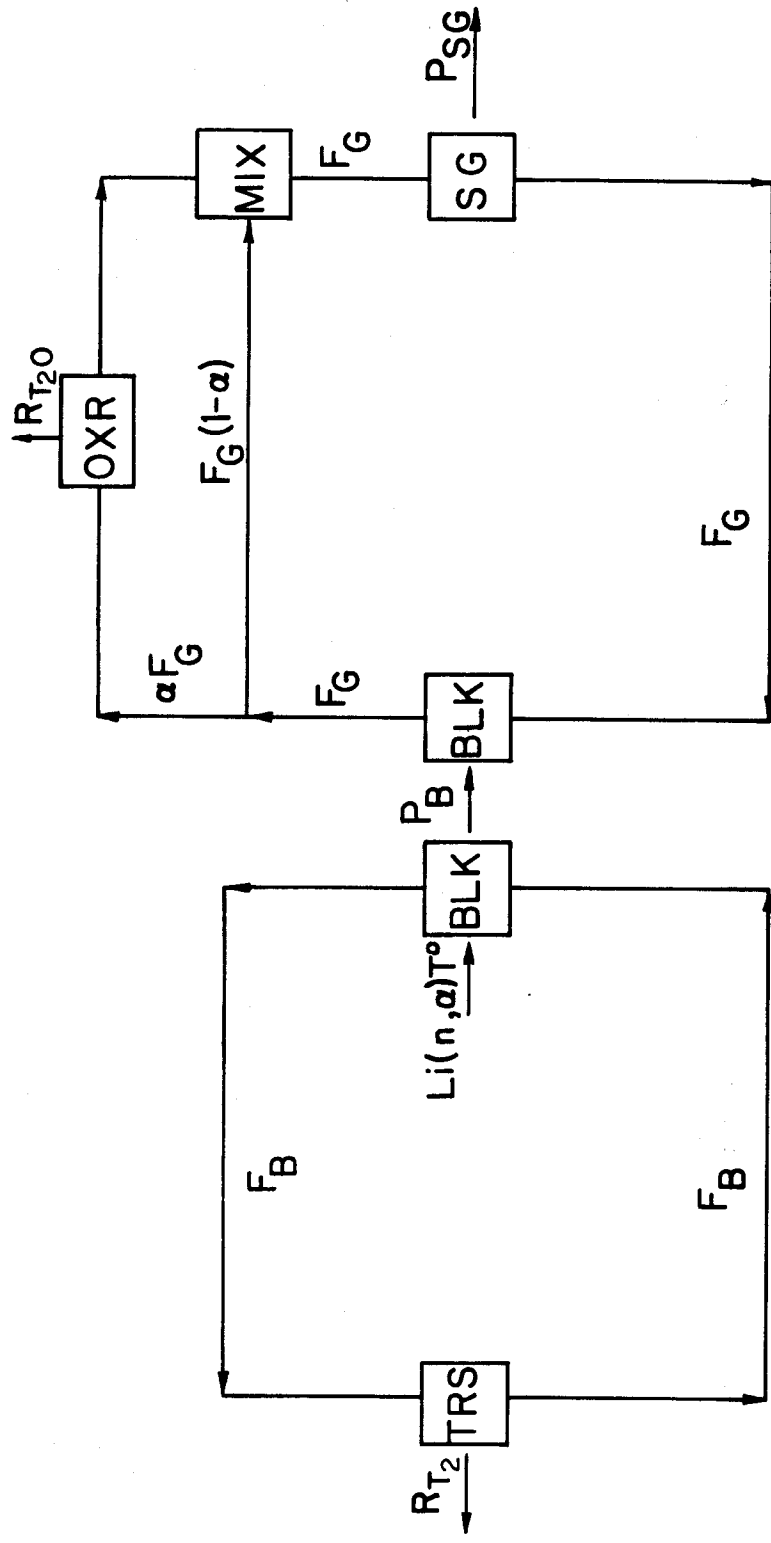
- Impulse corresponds to 77 Pa-s
- Steady state motion develops quickly ( $\sim 15$  shots)
- Steady state amplitude is less than 1% of length

## Planar Midspan Amplitude – Frequency Response



- Figure can be used as parametric design curve with scaling
- Maximum amplitudes occur at resonant rep rates (impulse period)
- Peaks for fundamental mode are highest
- Damping substantially reduces resonant amplitudes

# TRITIUM REMOVAL SYSTEMS IN LIQUID BREEDER AND HELIUM COOLANT CIRCUITS



## LIQUID BREEDER CIRCUIT

## HELIUM COOLANT CIRCUIT

$F_B$  = flow rate of breeder

$F_G$  = flow rate of coolant gas

$\alpha$  = fraction of gas flow to OXR

TRS = Tritium Removal System for liquid breeder

OXR = Tritium Removal by Oxidation & Adsorption

$T^o$  = rate of tritium breeding

$P_B$  = tritium permeation rate to the coolant gas

$P_{SG}$  = tritium permeation rate to the steam circuit

$BLK$  = Breeder liquid to helium coolant heat exchanger

$SG$  = Steam Generator

$R_{T_2}$  = rate of tritium removal

## ***Tritium Inventory***

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<b>Location</b>	<b>Tritium, g</b>	
<b>In Reactor Hall</b>		
<b>Fuel Supply (1 hr)</b>	<b>21</b>	
<b>Reactor Cavity</b>		
<b>SiC (INPORT tubes)</b>	<b>10</b>	
<b>LiPb (liquid alloy)</b>	<b>0.1</b>	
		<b><u>31</u></b>
<b>In Fuel Fabrication Building</b>		
<b>Fuel Processing</b>	<b>20</b>	
<b>Pellet Fabrication</b>	<b>60</b>	
<b>Pellet Storage (1 day)</b>	<b>500</b>	
		<b>580</b>
<b>TOTAL</b>		<b><u>611</u></b>